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# AVIATION WEEK

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Division of Sundstrand Aviation Ltd. Company, 8000005, SUNDSTADT, Western District Office, Hawthorne, Colorado

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1.



2.



3.



4.

## Which airplane has the Disc-Type Brake?

Here they are, a jet, a helicopter, a large modern transport and a brand early bi-plane. Which has the disc-type brake, the brand looking principle developed by Goodyear for the aviation industry? Ready with your answers? Here they are:

**1. REPUBLIC F-84F**—this fighter jet is equipped with Goodyear Disc Brakes of the Internal Type, great brakes engineered for fast, high-speed landings.

**2. SIKORSKY S-56**—this helicopter uses Goodyear Brakes of the Single Disc Type in two applications on the landing gear and on the rotor as well. You'll find Goodyear Disc Brakes on virtually every helicopter made today.

**3. DOUGLAS DC-7C**—this fast flying transport is a pioneer of something brand new in super efficient braking. Yes! They're Goodyear Disc Brakes of the new Tri-Motor Type—equipped with the highest energy absorption capacity per pound of metal jet produced. Especially designed for high-speed, weight-saving applications.

**4. FORD TRI-MOTOR**—this famous aircraft has a history of great service stretching back over

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In short, all of the aircraft shown here rely on Disc-Type Brakes built by Goodyear—in the majority of the aircraft throughout the world.

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# Defending Today's USAF Bombers . . .

# Electronic Armament From General Electric



**16 YEARS' SYSTEMS ENGINEERING EXPERIENCE BACKS B-47 AND B-52 ARMAMENT PRODUCED BY GENERAL ELECTRIC'S AIRCRAFT PRODUCTS DEPARTMENT**

TODAY all Boeing B-47 and most operational B-52 Air Force Bombers are equipped with electronic defense systems designed and developed by General Electric. The evolution of systems engineered bomber defense since 1948 has brought about today's completely integrated armament equipment, remotely operated by a single Air Force gunner.



1948 General Electric entered the bomber armament field with wingless turret drives for the Convair Aircraft B-36 Liberator.



1948 Boeing B-52 Superfortress was defended by the first remote-control armament system including General Electric remote-control turrets.



1948 G-E control and turret linked with a periscopic sight allowed Douglas B-36 gunner to aim and fire both upper and lower turrets by remote control.



1948 Standard equipment on the Boeing B-52 included an advanced General Electric, remote-controlled armament sighting system.



1948 Convair B-36 is armed with a complete G-E defense system including turrets, sighting stations, radar control and a G-E electronic computer.



1958 The Douglas B-48 was the first tactical bomber to be equipped with radar-directed, autonomous and armament system.

**PROVIDED:** Expanding electronic bomber defense growth at GE are covering applications for your Convair C-60, Air Force Products Department, Defense City, N.Y.

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**G-E TURBOPUMPS ON THE B-52** are driven by air ducted from the jet engines. These units supply the hydraulic power necessary to operate the bomber's steering, ejection seats, landing gear, bomb bay doors, and wing switches.



**G-E TURBODRIVES ON THE B-52** also operate on air ducted from the engines. These constant-speed drives each turn a 60-hp, 400-up alternator which produces electricity for the B-52's radio, lighting, instruments, and control panels.

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The Boeing B-52 Stratofortress, greatest Air Force jet bomber, uses General Electric turbopumps and turbodrives to help supply its hydraulic and electric power. Clocking up more than 100,000 hours operating time, these weight-saving power packs have helped to make the giant bomber one of the most advanced aircraft in the world today.

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a turbine, gearing, and integral lubrication and control systems. Because their basic design requires very few moving parts, G-E turbopumps and drives are engineered to give long, dependable service under the most stringent operating conditions.

**COMPACT AND SELF-CONTAINED,** these units are located near the point where power is needed. Such versatile, separate locations reduce "drag" by cutting engine nacelle size, and eliminate the weight of long

## Weight, Permit Increased B-52 Performance

transmission lines. The lighter system weight obtained with G-E turbopumps and drives permits increased aircraft range, speed, or payload.

**TOP RELIABILITY OF AIR-TURBINE DRIVES** is achieved by cross-manifolding the plane's air supply ducts. Through this method, all drives aboard the aircraft can be operated on air from one or any combination of engines.

**FOR DETAILED INFORMATION** on how G-E turbopumps and turbodrives, manufactured by General Electric's Aircraft Accessory Turbine Department in Lynn, Mass., can answer the accessory power needs of your aircraft, contact your General Electric Aviation and Defense Industries Sales Office, or clip coupon for free descriptive bulletin.

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**Wings of old take on a new turn in being true VTOL. Right closer.** Vertol research now takes on one of the most advanced assignments of the decade — high speed aircraft with vertical takeoff and landing capability.

**Project:** Vertol is now developing aircraft types which will fly as fast, as high, as far, as today's transports yet are able to take off and land at 0 mph forward speed.

**Qualifications:** Vertol's long-established leadership in the helicopter industry, Vertol's impressive research and development record in all aspects of vertical flight.



**Progress:** Vertol is currently developing a flying test bed for the Army under contract with the Office of Naval Research. In this aircraft — a small-scale version of the larger, more powerful production types to come — a gas turbine drives rotor propellers housed in the wings of the aircraft. To takeoff and land vertically, both wings and rotor propellers pivot straight up; the rotor propellers furnish the necessary lift. At altitude, the wings flip over into "normal" horizontal position, as the aircraft gathers flying speed, cruise in conventional and fast.

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*Aircraft Corporation*

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Soon, the skies around Canada will be patrolled by the first aircraft developed specifically for Canadian maritime reconnaissance duties...the Canadian-designed CL-35.

Developed from the British Britannia class by Canada, this is the largest aircraft ever to be manufactured in Canada and we at Canadair regard it as a tribute to our capabilities that the RCMP selected us to do the job.

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CANADAIR HAS PRODUCED MORE JET AIRCRAFT THAN ANY OTHER CANADIAN MANUFACTURER

## Washington Roundup

### Democratic Challenge

The Democratic National Committee has inserted Congressional testimony by Sen. USAF. Now, a strong officer in a major position to indicate Republican Party platform plans defending the administration's defense program over the past three years.

These are the Republican plans and the Democratic counterplans.

• **Republican:** "We have the strongest striking force in the world...which must continue to include (A) a jet-powered, long-range Strategic Air Force and a Tactical Air Force with a striking capability, superior to any other; (B) the most effective ground and ballistic forces; and (C) a modern Navy with a powerful naval aircraft arm (projected to keep the sea lanes open to meet our foreign needs); (D) an Army whose mobility and self-sufficiency are without equal."

• **Democratic counterplan:** "The Air Force Vice Chief of Staff Gen. Thomas D. White has said that in airplane after airplane that the Congressmen are approaching in its quality and surpassing in its quantity, and Air Force Chief of Staff Gen. Norton T. Timm told a Senate committee that it is so quality that we have been sure to stay ahead. However, the Soviets are now closing the quality gap." (B) Russia has already tested an SS-9 missile, a 1,500-mile missile by the end of the year, the U.S. has only recently finished a crash program to catch up with the Russians in long-range missile development, but according to the former Assistant Secretary of the Air Force Test Division, the program is lagged down in administration and type. (C) Vice Adm. Robert P. Burke, Deputy Chief of Naval Operations, says the Russian Navy will surpass ours by 1975 unless we embark on an elaborate building program. (D) Gen. Matthew B. Ridgway says that at the time of his resignation as Chief of Staff, an adequate missile force was not being built."

• **Republican:** "We have suggested and will continue to support an effective and well directed program of research and development."

• **Democratic counterplan:** "Lt. Gen. Donald Pitt, head of USAF's research and development program, has said that since 1955, funds needed for USAF research and development programs have been inadequate."

• **Republican:** "In this age of weapons of unacceptable destructiveness, we must not neglect the protection of the civilian population by all known means, while at the same time preparing our armed forces for every contingency."

• **Democratic counterplan:** Gen. Earle E. Pritchard, head of our Continental Air Defense Command, told a Senate committee in 1960 that if a Russian air attack were launched against us today, most of the bombers would get through, stating "I do not think our defenses are good enough."

### Public Relations Problems

Confidential sources have been circulating among top-level aircraft industry public relations directors regarding exploitation of other methods of handling the Aircraft Industry from public relations programs than the current contract with JEC & Kohnen public relations agency.

Many were surprised by one of the top industry public relations experts and has been circulated only among the public relations director of a few large aircraft firms whose preferences are less likely to be the direction of AIA.

At the AIA Board of Governors meeting in Williamsburg, Va., last spring, a substantial increase was voted in the full & Kohnen budget for the AIA program and several additional have been made in the agency staff in Washington as a result.

### Less Manpower for D. C.

Without further, the Defense Department has issued new instructions ordering the military services to cut back on personnel in the Washington area. The policy is to reduce the payroll to the lowest point possible, while still maintaining a capability to do the job.

Secretaries of Defense, Navy and Air Force have been ordered to implement the policy, and without returning to the use of temporary duty or change of destination to achieve its intent. Monthly reports must be made to show the number of military and civilian personnel on the job every 15 days.

### Pentagon Brotherly

Passage of Congressional campaigns will give Pentagon officials a headache before being called before the House Information Subcommittee to compare testimony on public information and service policies and practices. Although the subcommittee considers the Defense Department most important of all government agencies in weapons, it has developed plans to conduct hearings into the month.

Members want to concentrate on trying to get re-elected. Regardless of whether they're successful, the hearings will be completed sometime between October and the fall of the year. Subcommittee Chairman is Rep. John E. Moss (D-Calif.). Members are Reps. Don't Fausch (D-Ill.) and Glen H. Hansen (R-Meb.).

### Nike-Talos Evolution

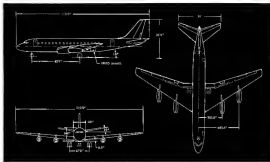
New it can be told. Direct result of the "Revolt of the Colonies" and Pentagon paradoxical dispute over development and control of subsonic missiles was the appointment of a committee to weigh relative merits of Gen. V. Mac and Gen. Mac. Following Defense Secretary Wilson's "initial Pentagon loss" - pass on former (AM-16, 35 p. 38). Gen. V. Macpherson, his special interest for guided missiles was made chairman of the sub-group to study the subsonic, convertible Bolo-type development into a new missile to be used in the air, evaluation and not include a dual between the two missiles, as demanded by a Senate committee.

### Co-pilot for S-58?

Civil Aeronautics Administration was called last week to make a co-pilot installation in a new version of Sikorsky's new S-58 helicopter. The request came from Air Line Pilot Assn. President Clarence N. Smith who told the CAA that an one-man operation of the S-58 "would present a potential hazard" because of the complexities of the new helicopter.

The CAA already has contacted the S-58 from an engineering standpoint, but a decision on crew requirements will be included in the operating certificate issued to New York Airways, which plans to start flying the helicopter sometime this month. CAA will decide whether the S-58 needs a co-pilot after studying the N.Y.A. route qualification tests.

—Washington staff



**DOUGLAS** DC-9 three-view drawing shows exterior is nearly identical to that of the DC-8. Dimensions are those of DC-8, indicated by roughly 8 scale. Dual wheel main gear differs from the quad wheel in the larger airplane.

## Douglas DC-9 Performance Is Revealed

**Top cruising speed is 503 kts, at 30,000 ft., payload is 21,340 lbs., and range is 2,425 nautical miles.**

Los Angeles-Douglas DC-9 medium range jet transport has reached the point of a final design.

The DC-9 is intended to compete with the Embraer 90B (AW July 15, 1976, p. 40). In general dimensions and thrusts it appears to be a replica of the DC-8 driven by approximately 8 kts. The 90B is somewhat larger, having a maximum gross weight of 175,500 lb. compared to the DC-9's 154,000 lb. maximum.

Consent has not by airplane comparable to the DC-8.

### At 85% Power

Top cruising speed of the DC-9 will be 503 kts. at 30,000 ft. with a gross weight of 150,000 lb. using maximum cruise thrust which is 85% of overall rated power. The speed limited payload has been set at 21,340 lb. of passenger and cargo.

Range will be 2,425 nautical miles with 77,000 lb. of fuel, takeoff gross weight of 154,000 lb. and an average cruising speed of 495 kts. at 30,000 ft. Block speed ranges from 190 kts. for a range of 300 nautical miles to 445 kts. for 1,000 nautical miles to 600 kts. for

3,000 nautical miles. Design level flight Mach number above 29,500 ft. is 0.84.

### Powerplant Choice

Power for the DC-9 will be supplied by four split compressor Pratt & Whitney JT5 turbojet engines, rated at 7,700 lb. thrust each. The engine is a true axial-flow of the New England 352 and was originally intended as a low specific fuel consumption, small thrust engine for use in fighters. The General 352 will use the 18,000 to thrust General Electric JT9 turbojet engine.

Total usable fuel capacity of the DC-9 is set at 15,578 lb. For one with the JT5 engine, Douglas has specified 74 middle cut gasoline MIL-E-5772 aviation gasoline and Pratt & Whitney specifications PW4-122 kerosene. Pratt & Whitney are now testing approval from the American Society for Testing Materials (ASTM) for general use of PWA 522.

Landing field length specified for the DC-9 without thrust reversers is 3,400 ft. at 5,570 lb. per sec. landing gear weight with a landing weight of 165,000

lb. Douglas has made no claim regarding landing field length with reverse thrust. Thrust must be equal to 40% of wet forward thrust to meet company requirements.

Each engine will weigh 100 lb. Noise approvals will be used which weigh 15 lb. each.

As to the DC-8 and the Boeing 707, (AW June 21, 1976, p. 61) two sets of engines will be used, one pair in front and one outboard. The inboard engines are very small and give adequate control only at high speeds. The larger outboard engines are controlled through a preselected spring linkage. At high speeds, the spring linkage yields to

### New Boeing Transport

Boeing Aerospace Co. will soon announce an entry into the world jet transport field with a number of new types for a smaller version of the 707.

The aircraft probably will be fitted with the General Electric CRJ650, a modified version of the J7A or the Boeing 707. Douglas has said Boeing will have a design to have a top speed of over 600 mph.

high air load, and at speed increases it programs the washes out the effect of the outboard ailerons while retaining that effectiveness at low speeds. Thus it is to avoid adverse control at high subsonic Mach numbers. The Boeing variations differ in that the outboard ailerons are driven by a linkage which is automatically broken when the flaps are actuated.

### Wing Flap Spoilers

Like the larger Airlines get transport, the DC-9 will have wing flap spoilers in the upper leading edge of the wing to disturb lift in the airplane handles down in landing. A narrow-slat spoiler in the main cabin shock strut will prevent the use of the spoiler in the air. They are not intended to provide lateral control as are those of the Boeing 707. Spoiler brakes are located in the wing root fillet.

Instrument panel on the DC-9 is expected to comply generally with Aerobank Standard 275 of the Society of Automotive Engineers' S-7 Committee on Cockpit Modernization.

The engine order used in the DC-9 will probably be an advanced version of RCA's AVQ 10 with a thrust tube provision to provide a thrust height enough to be used easily in descent without the aid of a hand.

In its present design stage, the new engine has seats for 74 passengers (see above). As noted, final passenger capacity and accommodations will vary according to seating in the cabin of passengers. A cabin altitude of 6,500 ft. at an engine altitude of 40,000 ft. can be maintained by the cabin supercharger turbocompressor, which will be driven by engine compressor bleed air.

## Orval Cook Considered As New Head for AIA

Washington—Orval Cook, retired USAF general and a retired expert in aircraft design, is being considered by the Aircraft Industries Assn. Board of Governors to replace David C. Ransom as AIA president.

Cook, who was deputy USAF chief of staff for material, director of procurement for Air Materiel Command, and commander of USAF Logistics Group in Europe, recently retired with the rank of full general and is now living in Indiana.

MacKinnon, Gen. Joseph McNamara, president of Convair, and AIA chairman, confirmed Assistant Secretary's conference story of August 27 indicating that an actual loss is in for a successor to Ransom. In a statement prepared and distributed by Aero-Mexico, public relations director of AIA, Gen. McNamara said:



### C-130 Demonstrates Climb

Step climb after takeoff by a Lockheed C-130 Hercules transport is demonstrated by Maj. Gen. Albert V. F. Anderson, Jr., commander of the Warner Robins Air Materiel Area, and his staff. The aircraft was made with the engine loaded almost to its 100,000 lb. design weight. Specifications of the C-130 specify a 21.8% climb angle of climb on takeoff will bring the airplane to 21,000 ft. in 40 sec. (AW July 16, p. 72).

"The Board of Governors is even more deeply struck with the strong leadership which Maj. Gen. Ransom has given the Aircraft Industries Assn. during his term of office," Admiral Ransom has devoted himself vigorously to a thousands of members of military and civilian aviation, and has represented as a highly capable and efficient manager.

"Admiral Ransom has indicated that he does not plan to seek extension of his contract beyond his expiration date, Feb. 1, 1978. This is based on his personal considerations involving his own health. Under these circumstances, our Board does not consider it in the best interest of the Assn. to continue to carry his heavy burden of responsibility."

"As a consequence, a committee has been appointed by the Board of Governors of the Aircraft Industries Assn. to screen candidates for his successor. This committee has been active with the full knowledge and cooperation of Admiral Ransom."

AIA staff sources advised that who ever was selected for Ransom's post

probably would join the organization before the year's end, presumably as an executive assistant for a period before Ransom's retirement. However, in this source, sources discussed the possibility of this procedure, indicating a term of the prestige and rather being sought would only consider stepping directly into the position. "In actual practice, it would be hard to cover the transition of Ransom's contract period."

### Convair-Astronautics General Contract Let

McNair Construction Co. of Los Angeles will build the 546 million Convair-Astronautics facility out of San Diego (AW July 26, p. 58). Work will begin at once on the base plant, involving of 200,000 sq. ft. of earth.

Two connected server buildings for administration and engineering, a common plant building, and supporting facilities will be provided for Convair-Astronautics, which will develop, manufacture and do preliminary testing of the Atlas ballistic missile and related projects.



**SINGLE BOLT** of electrically powered ammunition feeds all six barrels of General Electric C-130 Vulcan electric motor.

## 'Gatling Gun' Designed for F-104, F-105

By Everett Clark

**Aberdeen Proving Ground, Md.**—General Electric Co.'s new six-barrel 30-mm Vulcan aircraft cannon that will arm F-104 and F-105 USAF fighters and B-57 supersonic bomber, was officially unveiled here last week. Credit goes to around 7,000 hours per minute.

First picture of the Vulcan appeared in Aviation Week's Aug. 6 issue (p. 137). Vulcan's high rate of fire and high firepower is sought into 30 lb. per barrel result from application of principles first used 64 years ago in the Gatling machine gun. Fifty-caliber machine guns used in World War II fired about 400 rounds per minute.

The Vulcan was developed for USAF's General Electric and Army Ordnance Corps supervision. Air Force designation is T-171. GE is now developing an aiming device for use with the Vulcan.

The six-barrel cluster is related counter-clockwise (looking from the barrel end) by either external electric or hydraulic drive. Each barrel has its own bolt that holds the shell firmly from entry to ejection, reliably chambering double feed, according to General Electric.

A single bolt of electrically powered ammunition feeds all barrels. Rounds are pulled up by the 5 o'clock position, fired at 12 o'clock and ejected at the 7 o'clock position. Indicators are that enough shells will be stored under the barrel until a round is ejected.

General Electric, the Army and the Air Research and Development Command's Air Proving Ground Center at Eglin AFB, Fla., have put four models of the Vulcan through extensive tests. ARDC said that after more than two years of testing, its project engineers found the Vulcan program "very reliable."

The Vulcan has been tested in an F-104 test aircraft at Eglin and has been tested in the Lockheed F-104.

Although no production contracts have been announced, General Electric said its production capacity will be 25 Vulcans a month by the end of the year.

### What Vulcan Will Do

GE and USAF claim three advantages for the gun.

- **Superior discipline**, through barrel stability and reduced vibration. Clamping into the barrel with together clamp, called the "whip" of a single barrel, and firing and stored at a high clockwise center around of a multiple gun usually have. Combined round also eliminates barrel vibration.

- **Unprecedented reliability** due to reduced power needs. Each round is equivalent of the previously fired round, "dead" rounds are ejected automatically and cannot cause stoppages. The gun is extremely cleaner and less susceptible to corrosion than previously used guns. The four per barrel rate of fire also means easier operation.
- **Little maintenance** is required. Vulcan's rotary motion and subsonic velocity

has more longer life for parts and mounting. The gun can be field stripped for cleaning and maintenance and manufactured in less than 30 min. Complete, disassembly and reassembly can be accomplished in less than half a day.

Details on range, muzzle velocity, and exact rate of fire are still classified.

### Vulcan Weight

Weight is 200-300 lb. and length is 72 in. Model A had 770 parts, the present Model D has only 445. Part number, designed for 15-caliber ammunition, weighed 71 lb. per barrel. Later models include several features not on the original model—a muzzle counter actuator, booster take-off gun booster, finches and mounting and gun clearing assembly.

General Electric and the Air Force concluded after low temperature and high-altitude tests that the Vulcan is suited as well at temperatures down to and including -57°F as it did at room temperature when the gun heater is turned on at the approach site. The test was to show. The tests also demonstrated that parts not located prior to firing performed satisfactorily in standard laboratory air conditions and at high altitudes had no effect on firing or on functioning of the electrical connections, according to GE.

Another feature of the Vulcan is a half rate of fire for strafing. Bedford Engineering Co. of Oxford, Conn., assisted General Electric in tank design and Air Research and Development, Eglin AFB, Fla.,



**VULCAN (left)** and a 1905 model Gatling gun are demonstrated side by side.

Conn., assisted with tank manufacture. Development of the T-16, first of the Vulcan series, resulted from the Army Air Corps' solution in World War II that weight and rate of fire would be drastically altered by speed of supersonic aircraft.

In 1946, General Electric's Aerospace and Ordnance Systems Division at Schenectady, N. Y., was commissioned to develop an aircraft cannon under supervision of Ordnance Corps, Springfield, Army.

Army and General Electric engineers made a detailed study of every gun mechanism patented in the country and decided that Richard Jordan Gatling's multi-barrel machine gun principle, first patented in 1862, provided the best rate of fire.

The first Gatling was designed for

55-caliber ammunition. Later it was modified for both 30-caliber and 40-caliber rounds.

The 1905 Model Gatling used of the demonstration here for comparison to the Vulcan fired 303 ammunition. The Army did not desire the Gatling's electric and 1911.

Although the four-cluster on the 1905 model was turned by a hand crank, Gatling received a patent in 1862 for an electric motor that was mounted on the rear of the gun using and drove the gun as the electric motor used in 1860 rounds per minute.

In the F-104, the Vulcan probably will be fired through a gun port on the left side of the fuselage below the cockpit. In the Convair F-106 fighter, the Vulcan presumably will be used as a flexible mount in the tail.

## Defense Awards \$5,335 Million Aircraft, Missile Contracts in '55

**Washington**—Aircraft and missile manufacturers were awarded \$5,335 million in Defense Department contracts during 1955. The figure, announced last week, was 14.9% of all Postage procurement.

Topping the list of firms with military contracts was Boeing Aerospace Co., manufacturer of the B-52 intercontinental bomber, with orders valued at \$707.1 million, or 13.2% of the total. Close second to Boeing was North American Aviation Inc., with \$700 million, also calculated to be about 13.2% of the total. Third highest was General Dynamics Corp., which in-

cluded Convair. The corporation was \$760.7 million in contracts, 14.1% of the total.

In the new list of 108 largest contract winners covering the last calendar year, General Motors Corp. took the top-ranking defense contractor, topped by 14th place with \$582.1 million, or 10.9% of the total. However, as a subcontractor for the forward half year period from July, 1954, to December, 1955, General Motors will rank as the lowest contractor with \$2.5 of all outstanding defense contracts. Boeing ranks second with 46.6%.

In a statement accompanying the

list, the Defense Department pointed out that the 1955 awards represent a more normal period of procurement than the years of the Korean buildup—1950 to 1953—and three of previous years—1947 to 1954.

The Department said there are factors that tend to increase the present age of contracts needed to make the companies continuously rising the top 100 contractors. These are:

- Acquisition of subsidiaries that also are prime military contractors. A total of 104 companies, including parent and subsidiary firms, were represented on the list at December 1955, an increase over 248 of June 1955.
- Increasing emphasis on aircraft and missiles. These products as a rule can be given only to large companies with the capability to act as prime contractor or subcontractor.

Fourth-ranking corporation on the list of 1955 ranked inside a United Aircraft, \$187.4 million.

Others in the first ten are: General Electric, \$179.2 million; American Telephone and Telegraph, \$165.1 million; Ford Motor, \$171.1 million; Lockheed Aircraft, \$141.7 million; Curtiss-Wright, \$139.4 million; and Douglas Aircraft, \$120.1 million.

## AF Finances Shark In Douglas C-124

**Waltham, Calif.**—U.S. Air Force transports in Shark \$34.6 million (estimated) general aviation firm. The firm, now a subsidiary of Douglas Aircraft Co., is standard Douglas C-124 transport.

Shark can be modified in a matter of hours from a transport to a potential combat aircraft. The firm is the world's largest of producing C-124 aircraft.

One of the Shark models served in Oklahoma City when a C-124 for Douglas at the National Aeronautics Show Sept. 1, 2 and 3.

Air Force has been using C-124 transports regularly in delivering Shark models to the Air Force Tank Test Center at Fort Belvoir, Mo.

Single C-124 transport is complete Shark. The package is rolled aboard on its special ground handling dollies. Wings and vertical tail fin are removed by ground crew at the destination.

Models are loaded for delivery at Northrop's flight facilities at Los Angeles International Airport. Field assemblies of major components are assembled on the ground at the Shark center at the launching site.

Shark, designed to deliver a nuclear warhead over atmospheric distances, is 74 ft. long, 32 ft. in wing span and 16 ft. high. It has a gross weight of 17,000 lb. and a 1700-hp Pratt & Whitney T57 turbojet engine.



**NEW FETTER** details released by Navy confirm earlier Aviation Week estimate of 24 ft. length and 15 ft. wingspan (AVI, April 5, p. 71). Shaded gross weight is 5,020 lb. It combines pressurized fuselage and 100 turbulent postwing with a standard Navy release catapult. Active housing system guides the F-6 to follow normal torpedo range where a radio director gives the primary aiming signal. With automatic released, the torpedo under a second ship and comes in target. Launching is from a Lockheed P3V-7 Neptune capable of carrying two F-6s on underway pylons. Navy says a number of aircraft are in progress, and is planning to outfit squadrons for both Atlantic and Pacific fleets.

## Curtiss-Wright Orders Water Loop

Gladwin City, N. Y.—Curtiss-Wright contract to design, build and operate a water loop test facility for nuclear reactors has been awarded Walter Kiddle Nuclear Laboratories, Inc.

A water loop is a piping circuit in which the same fluid is circulated to carry heat away from a reactor core. In this case the laboratories are modifying a 50 ft. loop built five years ago for the Atomic Energy Commission. It is composed of an industrial-type pump and liquid piping from two to ten inches in diameter.

Wilbur E. Kiefer, president of the laboratories, explained that water could be pumped around this circuit at fairly high velocities and pressures. A 4-ft. section of 2-in. pipe at one of the reactor legs of the circuit will be removed after shut-out, be used, and in its place four elements which Curtiss-Wright would install will be connected. The contractors for these elements will adjust the accurate pressure and temperature taps for instrumentation.

Presumably the object of this loop is to test the cooling system which the laboratories are under contract to design for Curtiss-Wright's nuclear aircraft powerplant. They will investigate the heat transfer and the transfer of the circuit.

Another contract which Kiddle has with Curtiss-Wright covers the design of a liquid metal loop for Curtiss-Wright's Decheman, Pa., nuclear test

facility. This could either be used independently or in conjunction with the swimming pool reactor planned for Gladwin.

The Kiddle Laboratories which took the contract to complete work larger than its major atomic program, concentrate on handling nuclear problems on a consulting basis.

Besides the Curtiss-Wright water loop, the laboratories are designing a circulating water cooled test reactor for Pratt & Whitney and are studying nuclear power batteries for Elgin National Watch Co. In the nuclear aircraft field the laboratories are doing work for General and Glenn E. Martin.

## Douglas, Lockheed Decline AIA Miami Bid

Washington, 24—Major U.S. aircraft manufacturers have declined an attempt to take part in the Aircraft Industries Association's Latin American Aviation Conference in Miami, Nov. 14-16 (AVI, June 4, p. 77).

Neither Douglas Aircraft Co. nor Lockheed Aircraft Corp. will attend the meeting to discuss a co-ordinated military and commercial program covering present and future Latin American equipment needs.

Spokesmen for the two companies said their sales staffs have good contact with and indicated in both cases that

they have more important sales interests at the time. Two competitors of Douglas and Lockheed are active in laying plans for the Miami conference. They are Convair and Boeing Aerospace Co.

Charles H. Smith, of Westinghouse Electric International Corp. and chairman of the AIA Export Committee, and Latin American representative at the session will tell about these problems. U.S. industry and airline executives leaders will tell what can be provided to meet their needs. The spokeswoman will report present and projected needs and availability of equipment for military and naval aviation.

Requirements for civil aviation products including transports, utility aircraft, helicopters, engines and range instruments also will be discussed.

No list of speakers for the meeting has been announced. The program originally was planned for last February, but was postponed in order from AIA President Dwight C. Koenig.

## Curtiss-Wright Pilot Killed in F11F Crash

William M. Doyle, 34, test pilot for Curtiss-Wright Corp., was killed at Edwards AFB, Calif., last week in the crash of a Curtiss-Wright F11F.

The plane was undergoing engine altimeter tests when a blowout just prior to landing forced Doyle to land and set the aircraft down in the desert desert of the landing zone. The aircraft is powered by the Wright J65 engine.



**FOUR SPEEDY SPARROW** air-to-air missiles are mounted on missile carrying version of McDonnell Douglas, designated F4H-2M, which went into operational service last month. Deuce carries 23 mm. cannon as part of armament. The McDonnell J65 engine fighter is powered by an Allison J75 engine.

## USAF Service Centers Will Be Cut to Five

Three of the seven Flight Service Centers at Air Force bases in the United States will be dismantled and their operations will be taken over by the four other centers and a new one to be located. The centers, which provide such services as transmitting flight plans and providing in-flight advisory service on request, are being reduced to provide more efficient, consolidated operation.

The March AFB, Calif. Center will close early in October and the Hamilton AFB center will take over its area of responsibility.

Centers at Wright-Patterson AFB, Ohio, and Langley AFB, Va., will be closed and the new center at a base not yet designated will take over their area in 11 to 15 months.

## Defense Asks Copier Instrument Solution

Helicopter instrument flight is one of the problems thrown up for solution to the nation's aviation by the Department of Defense through the National Aeronautics Council at the Department of Congress.

The problems, cited at currently, limiting the effective operation of Army helicopters, are:

- Vibration level detectors, to measure and present a continuous recording of critical vibration levels in crew area.

- Helicopter instrument flight, reducing the instruments, techniques and aids to present the necessary data to operate in instrument conditions equivalent to those now flown by fixed-wing planes.
- Rotor blade stall indicator, to sense and present a continuous recording of the blade stall on the retreating blade.
- Coating for rotor or duct-type leading gear to have a low coefficient of friction on dirt and ice, excellent abrasion resistance, good adhesion to wood or metal, and which will not loose to ice, absorb moisture, rot or deteriorate.
- Electronic strobeoscopes for gear components in action, for comparing counts given off by worn parts such as bearings or gears with the normal count of new parts.
- Infrared-to-visual band inspection techniques, for checking structural parts.

Further information can be obtained from the council, Washington 25, D.C.

## Propeller Shipments

Plane manufacturing companies aircraft propellers shipped propellers and parts valued at \$57.4 million during the first half of 1956, according to the Bureau of Census and the Civil Aeronautics Administration. The shipments represent an 8% decrease over the same period of 1955. Figures listed below under other products and services include expanded and developed work by companies manufacturing complete aircraft propellers.

Period	Value of All Products Total	For Military	For Civilian	Other Products and Services
(000 Dollars)				
1956				
1st Quarter	\$40,342	\$21,520	\$6,400	\$12,393
2nd Quarter	45,014	25,479	7,857	11,676
1955				
1st Quarter	41,120	26,246	5,369	9,504
2nd Quarter	45,757	23,827	6,864	9,966
3rd Quarter	34,379	21,630	3,805	8,944
4th Quarter	46,455	25,054	4,393	12,134





**WHITE** phosphorus bomb explodes below Alaska B-17 as low level run over ground target still burning from previous strike.



**LAB SYSTEM** used by F-100 built in technique for height timing during low level dash to target in need with atomic weapons.



**NAPALM** drop by a large formation of Republic F-84s envelopes a ground target during a firepower demonstration at Eglin AFB.

## Toss Bombing, Napalm, Phosphorus Etch Firepower Patterns



**ON THE GROUND**, a tight formation of eight F-84s atom napalm against target of buildings and old automobiles.

## Research Expense for Electra Holds Down Lockheed Earning

Research and development expenditures of \$4.7 million, largely for the Electra turboprop transport, reduced net earnings of the Lockheed Aircraft Corp. from \$9 million to \$3.21 a share for the first half of 1975 to \$7.5 million to \$2.65 a share last year.

Lockheed's Manager, Services Division, which is flying two types of research contracts for the Air Force, is near the contract stage with both the USAF and Navy, but the new business potential could not be reflected in the first-half report.

Mainline sales rose from \$285,508,100 last year to \$263,845,000 in the first half of 1976. Lockheed's military work includes operating a machine tool center for the Air Force and nuclear reactor design and development contracts.

Lockheed backlog increased to a percentage high of 51.3 billion. Commercial backlog of orders for 210 airplanes also reached a new high of \$109 million. Commercial orders increased in the first half of 1976 totaled 52 airplanes from 11 earlier. However, first-half sales of \$51.9 million were down from the record \$66.3 million established in the first half last year.

Total sales were \$115 million, compared to \$172 million last year. Improvements in manufacturing efficiency produced a more favorable cost-to-sales ratio, and strengthened Lockheed's earnings position.

### United

United Aircraft board of directors have voted stockholders the right to subscribe to shares of new convertible preference stock at ratio of one share of new preference stock for each 10 shares of common stock on record at close of business on Sept. 17 or at such later time as the registration statement covering the new stock, becomes effective.

Free of new shares, to be determined by board of directors shortly before subscription offer is made, will not be less than the par value of \$10 per share.

### Solar

Net loss of \$189,200 on 27 cents a share was reported by Solar Aircraft Co. for the quarter ending July 31 as a result of a month-long strike at its Des Moines, Iowa plant. That the loss was temporary is indicated by net income of \$21 million obtained in the same quarter, bringing the company backlog to \$30 million.

Solar directors declared the regular 25-cent quarterly dividend. Sales also for the quarter, the first of Solar's fiscal year, were \$11.9 million compared with \$12.9 million the year before.

When the strike began, the Des Moines plant was engaged in high-volume production of jet engines and guided missile assemblies and the San Diego plant was changing over to new programs.

The result was a sharp cutback in anticipated sales and profits.

Solar net was \$180,600 or 36 cents a share in the comparable three-month period in 1975.

### Fairchild Canada

Fairchild Canada and Investment Corp. issued \$1.75 a share for the first half of 1976, including 42 cents from the sale of its facilities at Jamaica, N. Y. Net sales of the parent N. Y. firm totaled \$20 million, compared with \$16 million for the first half of 1975. New orders rose to \$13 million.

### General Controls

General Controls Co., Glendale, Calif., net was \$214,101, or 69 cents a share, compared with \$232,990, or

\$6 cents a share in the first half of 1975.

The first half 1976 sales were higher—\$13.4 million compared with \$12.5 million last year.

### B. F. Goodrich

Sales and net income were down compared with the first half of 1975 for the B. F. Goodrich Co., Akron, Ohio.

First half 1976 sales were \$364 million, net \$21 million and earnings per share \$2.41. Last year the comparable figures were sales, \$372 million, net \$32 million, and earnings per share, \$2.62.

### Dividends

Dividends have been declared by the following companies:

- Lockheed Aircraft Corp., 60-cent quarterly dividend payable Sept. 11
- Northrop Aircraft Inc., 45-cent quarterly dividend payable Aug. 31
- Chance Vought Aircraft Inc., 40 cents payable Sept. 24 to stockholders of record Sept. 7
- Glenn L. Martin, quarterly dividend of 40 cents, payable Sept. 16 to stockholders of record Sept. 7

## Royal Australian Navy Grounds Sea Venom

McDonnell-Eagles fleet of new seagulls to be grounded by the Royal Australian Navy following preliminary investigations of a fatal crash. Defects in the control system are suspected, and other trouble spots which are likely to reduce Sea Venom efficiency in certain conditions.

Australian Navy's new aircraft carrier, the HMAS Melbourne, is now all in new aircraft. The carrier is now all in new aircraft. The carrier is now all in new aircraft. The carrier is now all in new aircraft.

## News Digest

Royal Canadian Air Force announced increase of flying pay for its crew members from \$38 a month to a graduated scale of from \$77 to \$150 a month. New sales, effective as of July 1, is a addition to regular pay, and otherwise. Paymen to be paid in accordance with the terms of "the increasing complexity of military aviation in recent years."

Boeing KC-135 jet tanker input will continue from a peak of 17 to 20 a month, following USAF reconsideration that Defense Department allocate an additional \$150 million for the program (AW Aug. 28, p. 25). First of



## One airplane in a class all by itself

The exciting new Fairchild F-27 occupies a special place in the air world of today.

- the only twin-engine aircraft in production for local airline service
- the world's fastest executive plane—up to 240 mph cruising
- flies more smoothly, with less noise, and can climb faster than any twin-engine airplane now in airline service.

- the economical F-27 more than pays its way on short and long hauls... up to 2,000 miles.
- has unusually good short-field operation and engine-thrust performance.
- it's the least expensive aircraft in its class at \$2,000,000—costs far less to buy, to operate, to maintain.

There are no others in the F-27 class—it's one airplane in a class all by itself.

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**F-27** Friendship

THE FASTEST AIRCRAFT FOR AIRLINES  
CORPORATIONS AND MILITARY SERVICES

\*ENGINE: PUMP TO AIRBUS SPINOT EXCEL  
EXCELLENCE: PUMP TO AIRBUS SPINOT EXCEL  
PUMP TO AIRBUS SPINOT EXCEL

## Air Show Coverage

Airshow Week ended with the report from three major air shows in the Sept. 30 issue.

Publisher Robert W. Martin and Editor Robert Hertz will attend the Society of British Aircraft Constructors Flying Display at Farnborough, England. Managing Editor Stephen W. Jones will cover the Canadian National Air Show in Toronto, and an editorial team headed by Editor Robert Hertz will report on the National Aircraft Show at Oshkosh, Wis.

Other on-the-spot international coverage by the Aviation Week staff in September will come from Rome where Enzo Andolina, assistant managing editor (tech. info.), is a delegate to the International Congress of Astronautics and Astronauts where the International Air Transport Association will hold its 12th annual general meeting.

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KC-119s will make maiden flight from Renton, Wash., to Seattle around Sept. 10 for USAF delivery. Flight goes some even sooner if tests are completed.

Employment at Douglas Aircraft Co. topped 88,000 new production highs. Approximately 50,000 are at work on the three Southern California divisions. Total payroll for five divisions amounts to \$480 million a year.

Initial steps for installation of sales at New Zealand airports were announced by Civil Aviation Minister. Progress will cost \$2,000,000.

Bell Aircraft Corp., Helicopter Division announced sale of 38 Model 47G helicopters to French Air Ministry. Order is largest of year for division.

Gordon, Rostropovich, North Hollywood, received contract from USAF for joint cruise 25 ft. long, weighing less than 10 in. less. Cruise will be used by Flight Test Center at Edwards AFB.

Texas Aviation Corp. will manufacture additional McDonnell F4H Phantom wings and aft fuselages. New order is part of Texas' present contract providing for similar components for Navy fighters.

Republic of Indonesia ordered 555 surplus aircraft of UT-1. Although no new acquisition from the Government Aircraft Engineering Corp. First delivery will be made in 16 months.

Strategic Air Command's 77th Fighter Wing at Baguio, AFB will convert from Republic F-86s to the McDonnell F-101 Voodoos in October.

CAA Observer Tom Run is convinced that changes must be made in rules and equipment for commercial aviation under current rules. Current rules are too strict. Run, 47th Street, Chicago, Illinois, says that rules are too strict. Run, 47th Street, Chicago, Illinois, says that rules are too strict. Run, 47th Street, Chicago, Illinois, says that rules are too strict.

New electronic autopilots will be used in all Air Force Strategic Command aircraft. Contract starts with 13 advanced automatic flight control systems, the Lase 17, which uses functions in place of vacuum tubes.

HILL-1, Bell four place utility helicopter participated in test trials for Navy assault carrier. Trials off coast of Rhode Island. Single rotor helicopter is powered by 200 hp. Lycoming VO-635 engine.

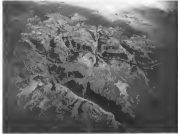
## Voodoo Tests Camera Capability



McDONNELL F4H and forward oblique camera on California to St. Louis. Right: Phantom was made at high altitude speed, with lighting through ground glass cockpit lens.



EDWARDS AIR FORCE BASE from 16,000 ft. shows runway, and pattern clearly.



GRAND CANYON now covers 1,500 sq. mi. Found the smoke is visible at upper right.







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## ICAO Pacific Pattern Revised To Permit 'Random' Flight Paths

Washington—A system of "random" flying over the Pacific Ocean that will give pilots a free choice of transoceanic air routes has replaced the historic fixed route pattern.

Under a new International Civil Aviation Organization ruling that became effective on Sept. 1, pilots can now plot uncoordinated transoceanic courses that will enable them to take full advantage of favorable winds and weather conditions.

The system is similar to that governing transatlantic flights.

Previously, transoceanic pilots were required to follow one of established routes between Honolulu and the West Coast. Controlled sectors to Wake and Guam were pre-designated tracks 50 mi. wide with a maximum altitude of 3,500 ft. Tracks westward to the Far East were similarly established to coincide with routes most commonly flown.

Pilots may now file flight plans over routes of their own choosing without conforming to the pattern or altitude pattern. Courses may be changed en route or flight plans covered only at designated fix points and only after securing permission from control centers.

All tracks are now set at a width of 140 mi. to allow a 50-mi. downwind and a 20-mi. buffer zone on either side of a control course. A vertical separation of 1,000 ft. is required, and aircraft must hold in a 38-mile longitudinal separation. (The North Atlantic tracks are 120 mi., 1,000 ft. and 30 minutes.)

Now in meeting space a minimum altitude of 1,000 ft. for transport aircraft to prevent interference with cargo activity, although it has agreed to give consideration to lower altitudes limits in certain cases. Airlines and, later, other consumers in one method of avoiding landings.

Chief advantage of the random system to the airlines lies in its freedom to select routes that offer the greatest asset from winds and which permit circumventing of poor weather areas.

The old system, based primarily on DC-4 equipment, was created in 1945 by the ICAO Northern Pacific Regional Air Navigation Conference at a meeting in Seattle after high altitude flight was set a barrier to aviation operations.

However, with the introduction of faster airplanes and pressurized cabins, the airlines felt that no satisfactory operational growth was opposed by the fixed-route principle. Presently, the airlines were anxious to take full advantage of the jet stream and other wind currents to maintain maximum operational and economic efficiency.

Furthermore, the airlines desired that jet transport operations will require greater flexibility in traffic flow than would be possible under the old route system. As a result, control of all air space to permit random flying was adopted at a regional air navigation meeting in Manila during November 1955.

All flights will continue to operate under the jurisdiction of Air Route Traffic Control Centers operated by the Civil Aeronautics Administration are located at Oakland, Seattle, Anchorage, Honolulu and Wake Island.

Control operated by the military are located at Guam, Midway, Kure, Ika, Okinawa and Tokyo.

### CAB Postpones Action On Bermuda Fare Hike

Washington—Civil Aeronautics Board has delayed action on an intercarrier fare agreement involving increased charges for discounts and Washington Bermuda fare. Such probably will be in effect by September.

The proposed fare plan was prepared at the request of the International Air Transport Association. The plan would offer a 40% discount from base fares to transoceanic fare Tancap.

Pointing out that it had disapproved similar plans in the past, the CAB said it would defer action on the new proposal until a new transoceanic discount.

The Board said it appreciates the desirability of facilitating intercarrier traffic but feels that this can now be more adequately accomplished through the use of charter services or other means, which do not involve the discount they have been provided.

The IATA agreement involving Washington-Bermuda fare would raise Eastern Air Lines' fares and Eastern's fares on the route to Bermuda when Eastern took over Colonial Airlines' fare.

Eastern's Colonial Airlines Domestic fare between Washington and Bermuda via New York. The current fare between Washington and Bermuda is the same as the New York-Bermuda fare. Under the IATA plan, the Washington-Bermuda fare would be increased by the amount of the regular Washington-New York fare.

The CAB said it would not mind a fare increase that reflected the mileage difference as the two routes (Washington to 34 mi. farther than Bermuda

fare is New York). But the Board found the proposed fare increase for purposes and would have an impact on domestic fares to the south and west of Washington.

### Northwest Awarded New Pittsburgh Routes

Washington—Civil Aeronautics Board has awarded Northwest Airlines authority to operate local service between Pittsburgh-Cleveland and Pittsburgh-Detroit.

With its new authority the airline can now compete for routes to serve as in the Pittsburgh-Cleveland and Pittsburgh-Detroit markets now dominated by Capital Airlines.

The CAB has completed an order which permits the New York-Chicago route, decided last September, with a decision to allow Eastern Air Lines to continue to operate between Pittsburgh and Akron/Canton, Cleveland and Detroit with a temporary restriction.

In the New York-Chicago case, the Board temporarily lifted a closed-door restriction that prohibited Eastern from carrying passengers between Pittsburgh and Akron/Canton, Cleveland and Detroit. Fuel diversion on the restriction was deferred for consideration with Northwest's application for increased authority.

Northwest was chosen over Eastern to offer new competition for Capital on the route because Northwest is the smaller of the two airlines. The Board said the process seems for choosing Northwest was that between Northwest and Eastern, Northwest is smaller, the weaker carrier and it is in need of strengthening.

### Mohawk Expects Profit From Mail Rate Change

Mohawk Airlines expects its 1955 loss of \$67,950 to change into a net profit after implementation of a temporary mail rate order which Mohawk operated during the last six months of the year. The local carrier name, in its annual report also predicts a substantial gain for 1955.

Mohawk flew 251,600 passengers a total of 51 million passenger miles during 1954, the airline reports, for a traffic growth of 20%. Mohawk's mail route is passenger flight along U.S. local service routes.

Describing the integration of Governor 1955 into its DC-1 fleet, Mohawk reports the Convair 440 superior in productivity, carrying capacity and characteristics to passengers. Maintenance delays were five to one over the DC in during initial operation. Mohawk reports that the airline looked out at the maintenance program being progressed.

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## Polish Airline Details Post-War Growth

**Warsaw**—After two years of scheduled operations, LOT's L-47s, Lotniew, (LOT), Communist Poland's state airline, reports that it has increased its passenger traffic fourfold over the pre-World War II level. It hopes to triple present traffic volume within the next five years.

Last year, the airline carried 134,822 passengers on its domestic and foreign routes, according to International Air Transport Association figures. The carrier itself claims an annual load factor of over 70%.

Comparing present traffic with pre-war Polish airline operations, LOT claims that using the 1938 traffic level as a base of 100, backloaders have reached an index of 550 for all services and 1,065 for domestic operations. Passenger traffic is set at 392 overall and 1,750 domestically.

The Polish Government established LOT after World War II with the help of the Russians as part of the "new" domestic infrastructure was required, and the airline began scheduled service in 1946. The carrier's schedule are tied into the transportation network so that flight arrivals and departures are timed to match railroad and bus schedules.

Flight frequencies reach a high of five to six trips a day on some Polish route segments, and air line service requirements regular airline schedules.

At present, the Polish airline is flying over its route network with a mixed fleet of L-1, D-12, B-14 and C-47 equipment. The carrier also operates dozens of light aircraft.

LOT's domestic route structure is designed to connect Warsaw with the main provincial centers—Kielce, Wrocław, Łódź, Poznań, Bydgoszcz, Szczecin, Lublin, Gdańsk, and Wrocław. The carrier's routes also connect the provincial centers and links the southern industrial regions with the northern airports.

Connections are made at Warsaw with international routes. LOT also flies to Paris, Brussels, Stockholm, Basle, Copenhagen and Vienna in Western Europe; to Moscow and to Belgrade, Sofia, Prague, Budapest and Bucharest in Eastern Europe.

Warsaw's Mroczek Airport serves as a perfect point for operations of such large carriers as KLM Royal Dutch Airlines, Aeroflot, German Deutsche Luft Hansa, Czechoslovakian Airlines, MALEV Hungarian Airlines, CVA Bulgarian airlines and TARDIM Romanian airlines.

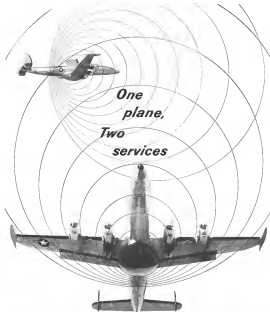
Along with its transport operations, LOT operates a fleet of light planes for agricultural, forestry and aerial mapping work.



INTERNATIONAL routes of LOT link Warsaw with European capitals—East and West. Warsaw's Mroczek Airport handles dozens of light aircraft, but also operates a mixed fleet of L-1, D-12, B-14 and C-47 equipment.



DOMESTIC route structure of Communist Poland's state airline, LOT, connects Warsaw with some provincial centers and links southern industrial regions with northern airports. Flight frequencies reach a high of five to six trips a day on some Polish route segments, and schedules are tied into the national transportation network so that flight arrivals and departures are timed to match railroad and bus schedules.



In this previous age when only one transportation weapon was dominant it was easy, the real test of maintaining the aircraft's fast line of aerial delivery has been obtained by the Lockheed Constellation built under patent plans—like U.S. Air Force's BC-121D and the U.S. Navy's WV-2.

In defense and fire, in five weather and fast, these dual constellations operate around the globe, at speeds up to 350 mph. Pressurized and air-

conditioned, they provide 75-degree extra comfort for a 21-seat crew, even during outside temperatures in 90 below zero.

The extensive delivery capability in the belly volume of these great planes was developed by Lockheed—just like the design to make sure an area of 45,000 square feet under floor, 30,000 ft. inside.

Only Lockheed produces long-range, AEW planes—a tribute to the all-weather capabilities of the DC-1-1049-WV-2 and to Lockheed's leadership in the day-to-day delivery systems field.

# Lockheed

AIRCRAFT CORPORATION, BIRMINGHAM, CALIFORNIA Lock is Lockheed for Leadership

## Shortlines

► **British Overseas Airways Corp.**, operating its New York-Bermuda service with a passenger load factor of 97% in May, June and July. BOAC operates two five-hour flights a week and a daily tourist service over the route, all with Viscount equipment.

► **European airlines** will increase flight frequencies by 7% and seat capacity by 13% under new schedules which become effective in October. Capacity will be increased with the introduction of Boeings, DC-8s, Metropolitan Viscounts, Major and S-16 equipment this winter.

► **International Air Transport Assn.** reports that 5,136,000 passengers traveled over the scheduled routes of the 71 IATA airlines last year. International passenger traffic increased 19% in 1955. While international passengers composed 15% of the total on all services of IATA airlines, they accounted for 29% of the passenger miles flown in scheduled service in 1955.

► **Kuwait Airlines** has selected Atlanta as a diversion base for its medium route. The base will handle eight scheduled flights a day, including a non-stop Atlanta-Chicago service established this month.

► **Swedish** has three two-week tours of European cities made for introduction of the IATA express bus on Oct. 1. The stopovers then include trips to London, Amsterdam, The Hague, Brussels and Paris for \$585, to Geneva, Madrid, Paris and London for \$485, and to Rome, Nice and the French Riviera, Paris and Versailles, London and Windsor for \$685.

► **Trans American Airlines** reports a 35.4% increase in gross revenue for the first half of 1956. Revenues for the first six months were \$7,084,478, and the airline flew 120,149 passengers 150,085,000 passenger miles.

► **Trans World Airlines** will now give ticket stock to firms that are frequent TWA customers. A company travel clerk checks reservation by phone, makes out the ticket and the company pays directly on the transportation card. TWA also gives a car fee right before clearing plans of the Chase Manhattan Bank.

► **United Air Lines** will shift its family plan schedule Oct. 1 as it will be offering from Miami, noon to Thursday noon instead of the present Viscount to Thursday schedule.

## AIRLINE OBSERVER

► **CASA** has agreed to employment requirements for traffic controllers from 18 to 100 aircraft per month to run the expanded service facilities scheduled under the five-year federal survey plan.

► **Air Line Pilots Assn.**, an affiliate of the Air Line Pilots Assn. and representative of some 1,100 office workers and station agents of National Airlines, is suing the company for damages and reinstatement pay. The union claims that failure to settle two weeks notice on lay-offs because of a threatened pilots strike was a violation of contract contract. Pilots had set August 17 as a strike date, and the airline showed operations to a virtual standstill in preparation for the strike which the pilots failed to carry out (AW Aug 27, p. 48).

► **Aeroflot** is flight testing the Tu-104 for scheduled service to replacing B-1E and B-14 equipment with the jet transports on certain specified trips. Tu-104 passenger flights have been conducted recently from Moscow to Copenhagen, Bucharest and Peking. For some time, the aircraft has operated without incident, giving new air over the Moscow-Orient Express route to Khabarovsk in Siberia (AW July 9, p. 38).

► **Air Transport Assn.** has received permission to intervene in the general passenger fare investigation by the Civil Aeronautics Board despite objections of the UAR Board General.

► **British Airways** signed British Overseas Airways Corp. to operate its Bermuda 10th over the North Atlantic route into New York to help boost sales of the turboprop transport BOAC, however, will schedule the Bermuda over Europe routes when the new aircraft goes into service in October. With the Bermuda and the Comet IV, BOAC plans to recover half ground on its Eastern and African routes from which approximately 65% of the airline's traffic is derived. During transportation will be kept on the transatlantic route until BOAC receives the long-range Britannia 300s sometime next year.

► **Trans-Canada Airlines** will begin a pre-later travel plan on Oct. 1 after long waiting in the down-payment system. The credit plan will be available for flights on all TCA routes in North America and Europe.

► **Four female co-pilots** are being recruited by a West Coast supplemental carrier because of the acute shortage of transport pilots. The carrier needs pilots qualified to fly DC-3 and DC-4 equipment in 50 crew complements but will accept the women to double as stewards. While this is the first such request that the Federal Employment Agency has received, calls previously have been accepted for women pilots in the executive's service.

► **Chicago Helicopter Airways** is the new name of Helicopter Air Service, the airline contracted to operate a scheduled helicopter passenger service in the Chicago area.

► **Consolidated-owned Trans-Australia Airlines** and Australian National Airlines have closed tourist class fares in a new war being waged against the low-cost, pre-war-owned Ansett Airlines, which operates Canberra 340s on coast, and B-707s on Trans-Atlantic operations. Victoria-Victoria Ansett will shortly introduce Canberra Metropolitan 440s into service.

► **Northwest Airlines** will redesign the interior of their aircraft next year when delivery of 14 DC-7Cs and eight DC-40s is accepted. New interiors, suggested by Douglas Aircraft designers, will emphasize new-style seating, seat fabrics and curtains. Window coloring in varying shades will prevail on walls and bulkheads. Exterior design will undergo only slight modifications.

► **Braniff Airways** has opened a low-fare market with 26 separate rates to accommodate over 200,000 transient passengers annually.





**IN THE MEDITERRANEAN SEA**—Participating in amphibious exercises of the U. S. Navy in the Mediterranean, men are Sikorsky HO4S helicopters of the Marine Corps. They are shown above with the escort carrier Shensu.

which carried 15 helicopters from their home base at New River, North Carolina. The training exercises featured helicopter vertical envelopment tactics.

## AROUND THE WORLD WITH SIKORSKY HELICOPTERS



**SAVED—A DANISH SEAMAN**—A Sikorsky HO4S helicopter from the U. S. Coast Guard Station at Salem, Mass., takes a Danish sailor from the merchant ship Peske Gao 35 miles off Block Island. The sailor, stricken with acute appendicitis, was lifted in the basket visible below the helicopter and was flown to a hospital ashore for surgery.



**SAVED—A U. S. COASTGUARDSMAN**—Two days later the same HO4S helicopter picked up a U. S. Coast Guardsman from the Nantuxet Lightship. Also a victim of acute appendicitis, the man was quickly and safely taken to the same shore hospital. Helicopters have rescued more than 50,000 people in all parts of the world.



### HELICOPTER HISTORY



#### FIRST SCHEDULED HELICOPTER SERVICE

On October 1, 1947, the world's first scheduled helicopter service was started by Los Angeles Airways, carrying air mail. One of the Dow's original Sikorsky S-58s, shown, is still operating and has flown 30,000 hours. Los Angeles Airways was a pioneer in night operations and instrument flying, and carried over 70 million pounds of air mail in its first three years. The two now carries passengers as well as mail and express in air fleet of Sikorsky S-58s.

**LARGER S-58s FOR NEW YORK AIRWAYS**—The first of a new fleet of 12 passenger Sikorsky S-58s has been delivered to New York Airways, the helicopter airline serving Greater New York and nearby communities in three states. Babcock Helicopters World Airways has ordered eight for its European helicopter service. Both lines currently use Sikorsky S-58s. The S-58s will be the largest helicopters to be in regular airline service anywhere in the world.



**SIKORSKY AIRCRAFT**

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# Scientists Tackle Jet Noise Problem

By George L. Christen

Cambridge, Mass.—There are three basic ways to solve the noise problem both by a large jet engine, use a quieter engine, install a muffler, or permeate the public to accept the noise, according to the annotated consulting firm of Bell Research & Newman, Inc., who are devising considerable time and energy to the matter. The assessment of commercial jet flight is leading to progress in the problem.

In their studies concerning the specific problem of how to quiet a jet engine, Bell Research has studied the system, "Considerable current engine design and the method of mounting powerplants is going along under the wing, there is now no apparent, reasonable way of quieting a jet engine," it is stated, "but there are ways without appreciable thrust loss."

Not is the noise problem posed by jet aircraft expected to quieting the engine. At altitudes of 30,000 ft., at cruising speeds, however, by now in the engine design to exceed that of the turbojet themselves, a situation which may call for a weight penalty is unavoidable.

## Sound Specialists

REN is composed of a group of specialists in all phases of sound absorption and control. They have tackled such varied and difficult problems as silencing the supersonic NACA wind tunnel at Cleveland, designing a large anechoic (non-echoing) room for General Electric, and doing the acoustical analysis for the installation of a new turbine room and housing station for the United Nations building. Usually, a jet's tail pipe should act as a

large, the air material of a sound hole, says REN. But a long, thin jet will inevitably produce back pressure and decrease thrust, so the problem remains a tricky one.

Substantiating REN's opinion that large jet engines noise suppression will be difficult is a statement made by E. R. Callaghan of the National Advisory Committee for Aeronautics, Cleveland, at the Second International Congress on Acoustics recently held at the Massachusetts Institute of Technology. Callaghan reported on research he had done with N. D. Staden on full-scale jet engines.

## "Greatly Alleviate"

"I think that it is becoming increasingly evident that we are going to greatly alleviate the noise problem caused by jet engines," Callaghan said. "There may be some cost in performance and increased structural complexity and weight. As yet, it would not appear that we have reached our goal of a maximum of 15 decibels reduction of the maximum sound pressure level."

Continued effort, however, should make such reductions possible. The conversion of our crude test devices to finished business apparatus can reduce development cost. Furthermore, a great many problems of a practical nature remain to be solved. At the same meeting, Alan Towne of the Department of Aeronautical Engineering of the University of South Wales, England, commented that, although the first group of commercial jet aircraft, which saw service in 1959 and 1960, may not exceed the noise made by present propeller-driven planes, he did not see that the noise suppression

noise suppression now under consideration would reduce noise sufficiently in plane engines yet bigger, more powerful and more numerous. He emphasized the need for more research and the need for a more thorough theoretical understanding of the problem.

## 20 Decibels Too Noisy

Ground noise made by a large jet transport at takeoff is 15-18 db above the accepted tolerance level, according to REN.

Dr. Leo Berard, president and chief, because neighborhood tolerance to jet noise varies with its intensity, frequency and duration, it is hard to establish an accurate threshold of what will and what will not be acceptable to the ear.

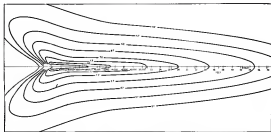
It was estimated by Ren as the corrected version of this paper given this year to the Institute of Aeronautical Sciences that the T-70, with two engines at takeoff power, will produce a maximum sound level pressure level of 135 db at 900 ft. (corrected) and approximately that independent studies made by the company of the Pratt & Whitney Aircraft J71 turbojet engine (this figure).

"Assuming an average airport and present-day neighborhood tolerance to airport noise, this level is about 20 decibels greater than currently permissible."

Noise made by the DC-6 series in plane engines is now being accepted by the urban as the standard against which the jet-T-70, DC-6 and Golden Aeronauts will be measured and which they will not be allowed to exceed. In a recent survey made by the St. Louis Civil Air Board (AW June 15, p. 107) of the noise level of several different makes of aircraft taking off from Revere Airport, DC-6B engine was 130-135 db, which supports Berard's statement.

REN officials cited these three methods of solving jet noise most acceptable to these proceedings who have taken to them:

- Use a less noisy engine. Such power plants as begun and low temperature jet are inherently less noisy than the more conventional design. The jet engine, such as the Bell-Boeing Co., is less noisy because the engine bypass air reduces the exhaust gas temperature, the engine, all the noise created by a jet engine at full thrust is caused by the exhaust along the nozzle



CONTOUR plot of jet engine noise is plotted along a scale marked both in miles and feet. Sound levels are in decibels.

between the making exhaust and supersonic exhaust air.

By designing a jet which would have a lower total turbine inlet temperature, noise can also be reduced because exit velocity decreases and tail pipe pressure increases, both being factors which help quiet the engine. Both types of engines will give a noise reduction of approximately 6 db with no loss to thrust or performance.

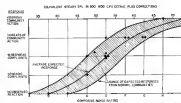
• Use a noise suppressor. Depending on the design, tail pipe noise suppression may be expected to cut jet engine noise by about 6 to 15 db. Drawbacks of noise suppression are that this will require a strong weight penalty and will create thrust losses ranging from 1 to 15%, both being undesirable in the aircraft. Another factor not to be ignored is that no jet engine muffler capable of doing jet noise.

• Influence the public. An aircraft, psychological factor, but one which may be as effective as the engine type or hardware noise reduction method. In an effective public relations campaign which will increase neighborhood tolerance of airport noise 10%.

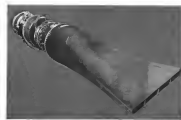
An unknown factor is whether the different trend character of jet engine noise compared with piston-engine airplane will result in a different neighborhood tolerance.

A trend which may mitigate the increased noise of jets is the increasing use of air conditioning, which serves as houses being closed both summer and winter. Ground windows and doors do much to reduce noise. This is even more applicable to office and other buildings surrounding the airport.

A noise problem peculiar to jet transport is the fact that, in general



AVERAGE expected community response to airfield noise as a function of noise ratings.

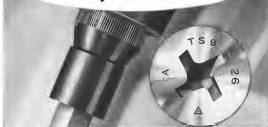


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One Watt	Automobile as Highway Noise-Shooting Noise-Compensated Level Noise-Very Soft Whisper
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**TOOTHED WHEEL** on one of the Pratt & Whitney T17-L turboprops powering the Boeing 707 is one of several modifications developed by Boeing engineers to quiet the greatest noise level of the four-pulsed propellers. This demonstration, conducted at a mock-up on a testbed built at Boeing Field, was held for the benefit of 50 airline and Civil Aeronautics Administration representatives. The 707 was started, tested and returned to the ramp and parked rapidly, during a five-hour period to simulate operation in today's air traffic. A Boeing KC-97 refueling tanker was towed to the simulated ramp and the second component of passenger and baggage handling gear was moved. Turboprops and velocity of the jet exhaust against the building walls were measured.

team, the "brake" of the boundary layer outside the "nose" of the engine at speeds exceeding 14,000 ft. and at sound frequencies above 600 cps. (Important speech range is about 140-1,000 cps.) At 40,000 ft., engine noise boundary layer noise will predominate completely.

Research cited these reasons for jet engine becoming quieter at cruise altitude:

- Less air (the air being used) is propelled through the engine, therefore, less noise is generated in the short phase.
- Lower velocity gradient exists across the short phase since the forward motion of the aircraft reduces the relative speeds of exhaust and ambient air by an amount equal to the speed of the aircraft.

It is harder to generate noise as altitude increases because the air "dies" far more easily when excited.

Consequently, at low altitudes and high engine speeds, the powerplant's roar will be the dominating noise, particularly in the rear of the plane during

take off and climb. R&D officials believe that it will take about twice as much soundproofing (in terms of weight) to quiet the rear of a commercial jet cabin than is required in a piston engine plane. However, the noise reduction may first solve an avoidable problem in quieting the interior of jets—the real problem lies outside.

Since, at thrust power, 100% of all noise produced by a jet engine is generated behind the engine, all muffling schemes are concerned with this noise and any sound created within the engine itself is discounted.

The noise is caused by turbulence occurring at the shear layer between the exhaust gas blowing through the still surrounding air. High-frequency noise occurs at the engine exhaust when the fan blades that lie between exhaust and ambient air are relatively narrow. Width of the shear layer increases as it recedes from the engine tail pipe and the frequency becomes correspondingly lower.

Noise power produced in the exhaust



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factor. Temco designed its rugged trainer to permit important up-to-the-minute changes with only minor modification.

Proving its worth in competitive flight evaluation tests at the Pensacola River Air Test Center, Temco's jet trainer was its Navy wings in June to begin testing the day when a cadet flyer's first solo will be jet-powered.

Engineer Openings in all phases of aircraft design and development: write to Joe Russell, Engineering Personnel, Temco Aircraft Corporation, Dallas, Texas.



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of a jet engine fitted with conventional, circular tail pipe increases with the 30% power of the tail section of the exhaust gas. Therefore, one slight increase in the tail pipe's perimeter, transverse to the axis of the engine, will result in an appreciable reduction in rear thrust of the low speed exhaust product between exhaust and ambient air. Ideal way to increase the tail pipe's perimeter is to fabricate it into a long thin section—a "thin exhaust" generally suitable in shape for the rectangular tailing edge exhaust fins developed for the very quiet Convair 440 helicopter. As Bernick put it, "We believe that an improved jet engine from the more standpoint would result if the exhaust — could be made in the form of a bag, thus do! Where this do could go on an exhaust, and meet all the other requirements of the exhaust and powerplant design is surely beyond me presently."

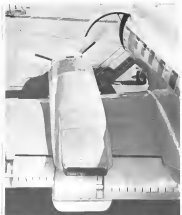
The do shape is more effective than the do shape configuration proposed by

being to the US earlier this year (AV) Jan. 30 p. 25), according to Bernick, because the more coming out of the lower rear exhaust tends to "sway," thus partially nullifying the effect of the doer.

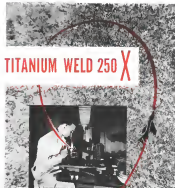
BN's acoustic experts have done so much work in quiet road travel that they have reduced the difficult science to an almost routine job. As they put it, "We have gotten to the point where we can calculate the amount of noise one can expect out of a given road travel."

A major accomplishment in the field was the noise control work, the firm did for the world's largest airplane wind tunnel, the 6 x 6 ft tunnel operated by the NACA at Ames High Pressure Laboratory in Cleveland. When the tunnel was first opened, its elevated power output means were so great that it could not be opened less than eight, while Cleveland couldn't sleep.

With the original soundproofing,



CONVAV 440's rectangular exhaust opening which replaces two circular jet openings (which is now almost impossible to pass). New muffler noise-reducing assemblies in first right row of 1000, application of acoustical tape to noise sensitive surfaces, addition of Douglas soundproofing are all part of sound lowering program for 440.



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the goal of a run get could be based 10 miles away. Neighborhood routes are quick and easy. After BBN's all-wheel drive control structure was added, a very reliable just the travel's cost could tell whether or not it was operating only by feeling the exhaust gas but not all time has been modified.

Currently, wind tunnel projects are under way for three customer lines for Douglas, and one each for General North American, Boeing, and Arnold Engineering Development Center.

In the same year, BBN has designed over 250 jet test cells for data on engine and exhaust manufacturers and for the USAF.

Another important facet of BBN's work is in determining the effects of aircraft noise on airport planning and on the surrounding community.

The firm has been solving the Port of New York Authority for several years on this problem. It still is making many surveys for the authority for such projects as locations of new housing near an air base.

The most serious of noise complaints is MIL speech interference level. It is a consistently increasing with speed, the public gets angry. The higher the frequency of interference, the more violently the public reacts.

Another big objection to noise (especially at a low community such as might have been quoted at New York where residents had reason to suspect aircraft noise with crashes and loss of life is disturbance at sleep.

As for the quality of the dollar machine room for GE, which measures 70x40x4 ft. and has walls 12 in. thick, BBN has done these new solutions joined in structure and some central couplings. For General Motors, which \$100 million (Toshiba model, quieted the latest Kennedy and Johnson outboard motor) provided some chlorine and water treatment systems for the experimental light and light. N. N. from, underlines some control studies on pre-hatched houses for U. S. Steel Corp., designed Diesel locomotive houses to sound like those schools, metal noise which required two large a volume of air to be effective).

The company also has done various structural design for such jobs as the Karpis Auditorium at MIT, the World Zionist Congress Building in Jerusalem and the Axi Magna Auditorium of the University of Venezuela in Caracas.

BBN was formed by R. B. Bell and L. Bernick in 1949. Both men, plus R. B. Norman and another member of the firm, J. J. Bernick, are on the faculty of MIT, besides managing BBN.

The firm grossed \$500,000 a year, mostly in consulting, it produces no hardware.



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*Lockheed P-38—equipped with a Honeywell Transmagnetic Fuel Gage*



*Martin XP5M-2 Sea Witch—equipped with a Honeywell Transmagnetic Fuel Gage*



*Lockheed P-46F Sabre—equipped with a Honeywell Potentiometer Fuel Gage*



*Martin EB-17B Canberra—equipped with a Honeywell Transmagnetic Fuel Gage*



*Cessna T-37B—equipped with a Honeywell Potentiometer Fuel Gage*



*Lockheed T-34C Starfighter—equipped with a Honeywell Transmagnetic Fuel Gage*



*Boeing B-47—equipped with a Honeywell Potentiometer Fuel Gage*



*Boeing B-47—equipped with a Honeywell Potentiometer Fuel Gage*



*Martin P-3H Hawk—equipped with a Honeywell Transmagnetic Fuel Gage*



*Douglas DC-7C—equipped with a Honeywell Transmagnetic Fuel Gage*



*Ford H-41—equipped with a Honeywell Transmagnetic Fuel Gage*



*Boeing B-47 Stratojet—equipped with a Honeywell Potentiometer Fuel Gage*



*Lockheed F-80A Shooting Star—equipped with a Honeywell Transmagnetic Fuel Gage*



*Cessna 441—equipped with a Honeywell Transmagnetic Fuel Gage*



*Grumman F-11F Tiger—equipped with a Honeywell Transmagnetic Fuel Gage*



*Boeing B-47 Stratojet—equipped with a Honeywell Potentiometer Fuel Gage*



*Lockheed F-80A Shooting Star—equipped with a Honeywell Transmagnetic Fuel Gage*



*Cessna 441—equipped with a Honeywell Potentiometer Fuel Gage*



*Lockheed C-130 Hercules—equipped with a Honeywell Potentiometer Fuel Gage*



*Douglas DC-7C—equipped with a Honeywell Potentiometer Fuel Gage*



*Cessna 441—equipped with a Honeywell Potentiometer Fuel Gage*



*Cessna 441—equipped with a Honeywell Potentiometer Fuel Gage*



*Boeing B-47 Stratojet—equipped with a Honeywell Potentiometer Fuel Gage*



*Douglas DC-7C—equipped with a Honeywell Potentiometer Fuel Gage*



*Cessna 441—equipped with a Honeywell Potentiometer Fuel Gage*



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writing the specifications to which prototypes are built. They are prepared for that engineering responsibility, both in a thoroughly up-to-date, broad technical education and by an industry job program which not only stresses loyalty to the company but apparently also matches grade, report or corps, and a sense of leadership and responsibility. Basic research is not the administrative duty of the universities or of the Ministry of Higher Education. It is primarily the responsibility of the Academy of Sciences of the Soviet Union, a body made up of the leading scientists in all fields of science and development which reports directly to the Council of Ministers and is assigned a large annual budget to sponsor research at a number of Academic research institutes.

However, the separation between universities and research institutes is more apparent than real, since the same man works in the various houses at the far north and in the afternoon reports research in the same field at a research institute.

The "academic hat" system in which leading academic personnel carry several responsibilities appears from the available techniques to operate with flexibility and effectiveness.

Scientists engaged in the leading world research which underlies the whole problem of flight do their work both at the universities and at the Academy-sponsored research institutes, where they are largely as research staff members. The design staff includes no research engineers and assistants or graduate students who act as part-time research assistants; the others spend part of their time as doctors or professors and part as research fellows or senior research personnel.

It is apparently almost a universal rule that the time of the academic personnel is divided between teaching, design and research. All in all, they form a reasonably well-organized group in the Soviet Union and enjoy high prestige and an excellent standard of living, with high earnings, excellent vacation pay and vacation at a designated resort and other advantages.

Compared to a world's rate of about \$30 million per month, the available pay scale for academic personnel is as follows:

- Civilian Institute or University-Academy: \$60-\$1200 (Doctor, 1500-1700 rubles, Professor, 1900-2100 rubles).
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In addition, there is a 15 per cent bonus at the individual level for the outstanding Soviet or only 15 per cent for the little Doctor Naik, and no extra

\$30 to \$50 rubles for administrative responsibilities for laboratory heads or professors in charge of a team. It appears from this table that the income of an academic worker is two to five times that of an industrial worker.

Furthermore, since it is standard practice to hold at least two jobs simultaneously, and the standard requirement (especially for senior staff members) are not strict, it is possible (and usual) to draw two salaries. A senior professor in charge of a laboratory might thus draw up to 7000 to 7500 rubles per month, and by combining and bookkeeping activities may double the total still higher.

### Salaries Doubled

The figures quoted above are as of 1948. Apparently they have all been approximately doubled since, possibly also in connection with the ruble inflation of the ruble.

Note the significantly higher pay rate of salaried scientists, an additional indication of the older place which they occupy in the Soviet scheme.

While scientists work on eight-hour workday basis, doctors and professors have a maximum work load of four to five hours and four contributions hours per week, the rest of their time being taken up by research, administrative work (including writing on money, medicine and government contracts), and consulting.

It is the rule for a doctor or a professor to prepare his own course rather than follow a specified textbook, step by step. This involves the preparation of notes which frequently end up by being published as textbooks. The practice of writing textbooks apparently approved by the state, is based on an appropriate sense of esteem for academic personnel and, in view of the great difficulty of publishing research papers, the best means of giving themselves known.

The present policy (at least in textbooks) appears to be to encourage the publication of a number of textbooks of slightly different viewpoints and the authors write in any subject tends to be quite large. (Five or six varied applied aerodynamics are known to this writer.) There is also much writing of monographs and treatises.

Review of the scientific or technical official information on the technical content and quality of the objectives of a Soviet research project is teaching and research in the subject just described, the documents which follow is limited to an analysis of certain of selected Soviet Soviet methods and some observations on the range of activities of graduates of a particular class, namely, the Chair of Fluid Mechanics.

Of the numerous textbooks contained

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in the course of this study, two were selected to extend the discussion further because they illustrate lines of research in theoretical fluid mechanics and because they are quoted often enough in the technical literature to indicate that they are in fact part of the technical baggage of most of the younger scientists. They are: Theoretical Hydromechanics by Kármán, Kibel and Buse and Mechanisms of Continuous Media by Landau and Lifshitz.

In general, in order to obtain the material presented in the two basic monographs a student must have an excellent knowledge of classical mathematical analysis, probably the equivalent of what an advanced graduate would obtain in the United States, a broad view of the line of analytical mechanics and thus application to a number of widely different physical sciences, and, last but not least, considerable sophistication and maturity in the handling of his mathematical tools.

Specifically, Theoretical Hydromechanics contains in great detail the principal mathematical methods of use in fluid mechanics: functions of a complex variable; expressions in orthogonal functions; method of characteristics; asymptotic methods; and so on. Each technique is illustrated by extensive application to one or several selected problems of practical interest which are treated thoroughly enough to allow comparing applications.

Mechanics of Continuous Media shows the range of applications of the science of fluid mechanics by presenting typical problems in the fields of structural mechanics, hydrostatics, and chemical engineering; in meteorology, geophysics, oceanography, geophysics, astrophysics, low temperature physics, and chemistry. This extensive, specific, practical detailed treatment of any single subject but gives the reader an insight into the uses of the field which underlies all the applications and suggests what fruitful analogies and even so which techniques can be grafted from one application onto another.

Certain solutions for expensive flow past cases, first studied in Germany in 1941 and in the United States in 1940 for instance, were noted in the Soviet Union as simple applications of a solution of a previously problem involving a completely different physical background (Sourin and Scholzer—1952).

It should be noted that both skill in mathematics and a broad outlook in classical mechanics have been part of the Russian scientific tradition since, well before the Revolution and were applied with success to many problems in fluid mechanics and aerodynamics by N. E. Zhukovsky (1847-1921) and his principal students, G. G. Glagolys (1890-1942). Both of these men

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reached maturity and occupied positions of prominence in the Imperial air safety system, and these IBM had the honor of sponsored research and development in Russia before and after the Revolution. The tradition of scientific skill and creativity is passed to the present day in the Soviet Union, as some examples will show.

### Three Men Selected

Among the senior teaching and research personnel, three men may be selected as typical of the Soviet effort in theoretical fluid and gas dynamics: L. I. Sedov (born around 1895) is an applied mathematician with a lasting

technique. He was connected with IACG in the 1930's and has been a professor of mechanics at Moscow University since the late 1930's and a member of the Academy since the 1940's. He has made significant contributions to the theory of the lift of wings (1932-1937), of ship wave resistance (1937), and of the impact of fluid bodies on a free water surface both singly (1951) and with several students (1950-1951). When high speed flight became a practical prospect, he attacked the aerodynamic problems which it raised at first with limited success (1947-1949).

In 1942 he gathered a group of young scientists who, under his guidance,

worked out the general theory of supersonic flight of thin wings and bodies (1941-1946). The last more than a decade, without an intermission (1946) and has undertaken a searching study of dimensional analysis and similarity in mathematics and physics, applying his results to the problem of very strong blast waves (1949-1952). He was appointed chairman of the Academy Commission for an earth satellite, not, as such, represented by the Soviet Union at the recent international symposium on satellites in Cosmology.

S. A. Christovovich (born in 1900), a star pupil of S. A. Chaplygin, did his last important work in 1935-1942, an extension of Chaplygin's calculations of gas jets so that it could be used to determine the performance of wings of new shape at speeds nearly up to the speed of sound. He played an active part in wartime research and has since taken the role of a senior consultant on fluid mechanics, a position highlighted by his recent election as chairman of the technical section of the Academy.

### Christovovich Papers

Christovovich has in the last five years presented or sponsored papers on subjects of interest in chemical engineering (Gas Flows in Pipes with Heat Addition and Supersonic Flows—1950), electrical engineering ("Problems related to the Construction of the Kuzbass Power Plant"—1951), mining engineering (Effect of High Pressure Fields on Rocking of Coal).

V. V. Golubev was born in 1904 near Volokolensk, where his father taught Latin in a parochial school. After a brilliant secondary school education during which he showed an interest in mathematics and the natural sciences, he entered Moscow University in 1925 and graduated in 1928 as mathematician. He then studied for three years in research, graduate study at the University, and teaching assistantships at several public and private high schools, and obtained his master's degree in mathematics in 1935. In 1937 he was appointed Professor of Mathematics at the University of Saratov. (Saratov in 1904 and later years expanded) where he became Dean of the Mathematics Faculty Department in 1945. Just then, as he was solving a classical dissertation on singular differential equations, all his domestic affairs were abolished (1949).

Circumstances thus forced him to take an interest in jet aircraft in 1945, and in 1944-1947 he began a study of lifting wing theories. Also 1930 he also taught mathematics at the Saratov Institute of Agricultural Sciences and he eventually published his notes "Principles of Mathematical Statistics Applied to Physics" (1950).

In 1940 he joined the research staff



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of TSAGI and was appointed Professor of Hydromechanics at Moscow University. In 1932 he received a permanent appointment as Professor of Mathematics at the Zhukovskiy Military Air Academy, and in 1939 he was given a notable rank as the Red Army. His simultaneous connection with the Hydromechanics Chair at Moscow University, the TSAGI main staff, and the Faculty of the Zhukovskiy Military Air Academy illustrates the "multitask nature"—which kept him in a well-informed and informed position as to all activities of the air development program. In 1941 he left the TSAGI staff and was appointed to the University of Stockholm, where part of Moscow University had been evacuated. In 1946 he became head of the Mathematics-Mechanics Department of Moscow University and a member of the Mechanics Institute of the Academy of Sciences. He died in 1954.

Judging by the existence of technical publications, it seems that visibility is also found in the new generation of Soviet fluid dynamists. In the periphery of her book, Wings of Frigate Ships is a considerable amount. R. A. Krasovskiy's book is a study in the field of the design of a group of rising wing sails to form a system in response wing theory. The three main authors of the group were P. I. Frenkel, an Aviation engineer from Soviet Union University with a thesis on self-propelled water skis, and M. D. Haidar, who had published a number of papers on ship aerodynamic dynamics.

## Younger Members

The younger members of the team included A. A. Nikolaev, G. I. Tsygankov, M. I. Gerasimov, L. A. Gorbunov, A. K. Krasovskiy, and A. A. Krasovskiy. After the completion of their task (and possibly the coming of some new aircraft) the group published a series of seven papers and two books in 1945-1947. Consequently, this group's first publications were part of the collection of the booklets on aerodynamics at Zhukovskiy's birth and the Faculty of the Academy of Sciences.

Solov's later career has already been outlined. Frenkel's book on integral form Frenkel, an applied mathematician as trained in aerodynamic problems, with the Faculty of Moscow University and on the research staff of TSAGI from 1933 to 1941. When TSAGI was transferred to Novosibirsk, he became a professor at nearby Tomsk University. After the war TSAGI headquarters was

located in the Moscow area, and he acquired the faculty of Moscow University. Since 1952 he has been a professor at the Khar'kov State University, his research bearing an international character in various streams and in some questions of the theory of relativity. Why a professor of Moscow University would "emigrate" for an appointment at the Khar'kov State University was a mystery until a reliable witness stated that the man's loyalty to the regime had been questioned after a friends' arrest.

## Hydrodynamic Flow Problems

Following, after completing his work in self-propelled machines, his income involved in hydrodynamic flow problems (problem of flight at very high altitude and very high speed). Haidar has returned to problems of ship dynamics, on which he has published a dozen papers since 1947. After his stay in Moscow he went to Nikolai, and he has been at Odessa since 1954. Among the various groups Nikolai has divided his time between extension of his own work on transonic flow and studies of explosion in gas media due to high pressure gas products. Tsygankov has not been heard from. Gerasimov has returned to Soviet Union. Gorbunov has applied some of the techniques developed at the Soviet Academy to problems of six-part dynamics, seepage and photoacoustics. Krasovskiy and Frenkel have not published any work. Krasovskiy's share is still unusual of with expensive wing flaps, of which he has given some interesting extensions. He and kept, the Soviet Academy aerodynamics task force has been dispersed.

The conclusion suggested by the long going career maintenance is that the education given in the field of theoretical fluid mechanics in the Soviet Union trains young people to take a broad view of their problems and to apply them in a number of physically unrelated situations. It also appears that it is the conservative policy of the Academy to limit the risk factor approach and to stand made in our time to form task forces of specialists in "launch the struggle" against whatever bottleneck is considered most serious to the overall research effort. In 1942 Solov's seminar constituted a task force of that time to attack supersonic flow, other task forces have been launched on the boundary layer problem (1938-1942) and jet engine design problems (1947-1954).

The task force approach tends to give a specialist's viewpoint opportunity to use the Soviet research and to emphasize immediate applications to short-range aims—the possible development of fundamental knowledge in itself. At least one observer (Black) suggests that so long as research in the West has enough lead to indicate what the significant problems are, the Soviet



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task, firms will be able to solve these problems as rapidly and effectively as Watson's group, but he questions the ability of the Soviet research apparatus to uncover and formulate new problems. Soviet performance is developing, but some serious shortcomings of the United States and Soviet scientific apparatus, independently of the United States, suggest that such a view may not be justified.

In any case, the availability of high quality scientific research personnel and the application of their skill to a given practical problem is most pressing task for an efficient and, for the present, generally successful plan of scientific management by means of what has been called above the "multiple hot system."

As a consultant to the design team, the Soviet scientist becomes familiar with the problem, as a member of an

Academy committee, he assists for the formation of a research group, as a professor, he is in a position to help recruit and direct a research group generally as well as under his chair. And the younger men have the task of the relationship between research and production and between research and education in the work of the task force and he becomes (not only) enough to lead one.

The Soviet method of managing scientific enterprises, which stresses variability in application, is particularly suited to theoretical investigations. It is much more to adapt a well-defined formula to fit a new physical problem than to develop an experimental technique. There is evidence that in the field of fluid mechanics much greater interest is placed on theoretical research in the Soviet Union than in most West-

ern countries because it is more and less expensive to develop the required staff and equipment and because, apparently, that approach is traditionally much more congenial to the Russian scientist.

Soviet progress in theoretical fluid mechanics is generally impressive. Soviet scientists have multiplied their full share of new ideas, and they are absent of these counterparts in most other countries. On the other hand, examination of technical and published papers on experimental research indicates that while the techniques developed in Germany, Britain, and the United States up to 1951 are known to the Soviet scientists, they have not sought new techniques and have used the imported techniques more for routine testing than to attempt to break new ground by means of fundamental experimental research.

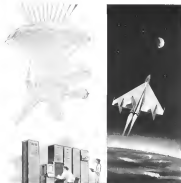
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graduate in carefully and thoroughly as possible, for, because of the somewhat mechanical nature of the average engineering education, the student carries a heavy responsibility both as teacher and as student research staff members.

The difficulties encountered in training enough graduate students to a sufficient peak of perfection are the subject of much concern at present in educational circles of the Soviet Union and, apparently, justly so, since the shortage of competent teachers is a major bottleneck in the expansion of the educational system.

### Aircraft R & D

Aircraft research and development are concentrated largely in the specialized research institutes—such as the Central Aero- and Hydrodynamic Institute (TsAGI) and other institutes of the Aircraft Production Ministry. But research done by the faculty members and graduate students at the polytechnic institutes and at the military academies (such as Zhukovskiy Air Academy in Moscow) is also a very important factor in the growth and development of Soviet competence in aeronautics. In a number of fields research task forces using the facilities and the leadership provided by polytechnic institutes, for instance, have contributed important results in the development of specific phases of equipment or in the solution of specific technical problems in the field of aeronautics. To illustrate this activity, we may consider the work of L. G. Lozinskiy, Professor of Mechanics at the Leningrad Polytechnic Institute.

### Mechanism of Drag

L. G. Lozinskiy is of the same generation and academic background as L. E. Sukhoi of Moscow University.

During the 1930s Lozinskiy was associated with TsAGI, taking a leading part in their studies of the mechanism of air resistance or drag. A German aerodynamicist, L. Prandtl, suggested in 1904 that a major portion of that drag is due to the viscous friction of the air in a very narrow layer next the solid surface along which it flows (boundary layer).

The practical importance of Prandtl's idea was just being generally appreciated about 1925 (possibly somewhat later in the Soviet Union), and it became Lozinskiy's task, first at TsAGI and after 1937 at the Leningrad Polytechnic Institute, to develop practical methods of calculating the flow in the boundary layer. Between 1941 and 1943 Lozinskiy and a group of students published a dozen papers to outline these solutions of the problem, which is an extension of a method proposed by Th. von Kármán in Germany in 1931.

Lozinskiy's method for the deter-

mination of being applicable to a wide variety of problems and of being quite straightforward and reasonably accurate. It is therefore well adapted to engineering calculation and appears to be in general use in the Soviet Union, but it is somewhat and not suited to the investigation of the effect of viscous forces on high-speed modern aircraft. This is not unusual. Lozinskiy is capable of more elaborate or fundamental analysis. The publication on technique and his booklets show that he has the solid mathematical education expected of a Russian scientist, but his boundary layer theory was meant to be, and within its range of

application is, an engineering tool within reach of the average designer. Since the war Lozinskiy and a large group of students have undertaken a program of basic studies of the gas flow in turbines. This problem is far too complicated for a manual solution. A good deal of engineering work and experimental calculation are required to bring out its essential features and to create an ideal model simple enough to permit analysis and solution enough to serve as a basis for engineering estimates. Recent publications have indicated that the Soviet aircraft industry has produced turbojet engines appreciably more powerful than those at present



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defense in the United States at the present time.

While knowledge and ideas from German and other sources have helped in this Soviet accomplishment, they would not have been divided and put to effective use so rapidly if entire groups, such as Lohmann's had not mastered the fundamentals of the problem independently and proposed a sufficient number of approaches to extend and acquire good Soviet ideas in an original and skilled manner.

In connection with each task it is essential to check experimentally the results predicted from various testing mechanisms, since these are supported by engineering intuition and must be discarded when they give reliable estimates. And, indeed, some computer assistance has been developed and used in connection with Lohmann's work, especially the later work on tailfins.

Although the Soviets are generally more watchful about their scientific testing equipment, they have recently described some optical equipment and some interesting prototype designs which indicate an ability to perform accurate measurements under difficult physical conditions.

The inevitable balance on expert talent is difficult to estimate engineering development task forces similar to the theoretical group described previously. The Soviet development experts are more closely linked with the equipment and the facilities, while engineering, even in a few leaders like Lohmann, cannot be compared with that of an applied mathematician. Nevertheless, the Soviet personnel at Soviet job technique institutes who can refrain to the study of basic engineering concepts and working models and equipment are doing their jobs with imagination, competence, and apparent success.

### Flow Techniques

An indication of the quality of the research done at the Stroukoff Academy is given by a recent booklet published by A. N. Zaka on flow visualization techniques (1958), which described the quality and type of equipment available there for experiments in aerodynamics. Zaka, who graduated from the Academy in the late 1940's, describes some of the wind tunnel and optical instruments available at the Academy and shows photographs of simple tests carried out there.

Facilities in which speeds up to twice the speed of sound are reached were evidently in operation in the late forties, and adequate instrumentation was also in use. The Zaka booklet and other data indicates that the Soviet facilities support the experimental equipment available at the Academy is kept fully up-to-date and in good condition.



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The work had, thus, strains of inequity, constraints, and intensely severe political supervision apparently take with a heavy toll of the students that only about a quarter of admitted college graduates. In one case where pertinent information is available, only six out of a section of twenty-two finished the course. Some wastage here appears higher than the average graduate school wastage. Since the only local communication is the Red Air Force, failure means assignment to a military post in the field.

These students who graduate receive portfolios of responsibility in the technical planning of the Air Force. In at least two instances, sufficient technical publications are available to form an objective opinion of the skills and training received at an air engineering academy.

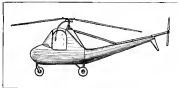
Professor A. P. Melnikov received his degree of Doctor of Technical Sciences.

from the Lombard Agr Engineering Academies (in 1942) with a thesis previously done under Professor Lantini in which he works on an original application of Lantini's boundary layer method to the problem of designing a wing of reduced air resistance. His last approach, concerned that work and produced six significant results in the Third All Union Applied Mathematics Congress in 1955. He is a candidate in computer applied research with great scientific aims.

Shortly after his graduation from the Zhukovskii Academy in 1932, A. N. Zolotarev published a textbook on Applied Aerodynamics which displays a thorough knowledge of developments in his field both in the Soviet Union and abroad, and the ability to organize his material clearly and to emphasize the points likely to be significant to the future; on the basis of that book one may also conclude that a thoroughly qualified aeronautical engineer.



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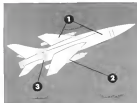
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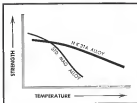
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## Space Navigation Challenges Engineers

By Peter A. Carmichael

Baltimore-Space, where the absence of atmospheric sensors that electronic rates would not be lost, devoted to atmospheric, optical or radar systems for the electronic guidance.

At the same time, the problems of guidance, navigation and communications for future space travelers, and energy splitting and fueling, for control and special communications in the event of space war challenge the electronic engineers.

At first glance, the problems of space navigation appear rather simple because the position of celestial bodies can be accurately predicted. It should merely be necessary to establish an accurate course from the base of departure to the future position of the particular destination desired, taking into account the various forces acting upon the craft.

However, in practice space navigation is made quite difficult by the extensive and complex required. A course must be in the most desirable regions, not only the exact knowledge of the position and trajectory of celestial bodies, which indeed is very accurately calculable but it also requires an equally precise knowledge of the position and calculable orbital direction of travel of the space ship and its velocity.

### Extreme Accuracy Needed

An idea of the accuracy required is illustrated in the following example: in making a voyage to the planet Mars it would be necessary to intercept the orbit of that planet at its nearest distance to the earth, which is at the radius of 36 million miles. For speed of travel of about 12 mi. per second, no error in launching angle of one second of arc would result in a positional error of the orbit of the trajectory of 500 mi. When one considers that the error of one second of arc is of the order of magnitude of error of the best theodolite under conditions of good observation, one can visualize the difficulty of maintaining a ship to any given trajectory within this order of accuracy.

A more likely event for a very accurate launching error is in the order of a minute of arc. This will yield a final error of 50,000 mi. Assuming that the ship is propelled with initial speed of 12 mi. per second, an error in initial velocity of 1% would cause an error at the end of the trajectory of about 500,000 mi. Even if the magnitude itself compared to the distance from

earth, would cause additional fuel consumption at the end of the trajectory in order to place the ship in the correct final orbit.

Early spacecraft will have no choice other than to follow very accurate paths due to the relatively low efficiency of chemical rockets but better fuels and engines will be developed to give a large degree of freedom in chosen flight patterns. We then envision two steps in the progress of space navigation.

• **Stage 1**—Chemical-powered ships cannot so accurately calculate their exploratory expeditions.

• **Stage 2**—Guided systems of regular space ships, and the advent of either heavy space traffic.

During Stage 1, space navigation will be almost exclusively optical, with the



### The Author

Dr. Peter A. Carmichael, the age of 31, was his doctorate degree in 1946 from the University of Göttingen, Göttingen, New York City, the age of a diplomat. Dr. Carmichael served in the British Royal Air Force during World War II, performing nightfighter missions without mishap from the age of 19. Three options for his war service ended at the time of his war end.

Returning to the U.S. in 1946, Dr. Carmichael joined Franklin K. Davis, Inc., the Glenn L. Martin Co. At the present time Dr. Carmichael is an advisory engineer in the missile-research department of the Glenn L. Martin Co., Baltimore, Md.

evolved existence of earth-based navigation help to the space ship, radar.

In Stage 2, we will witness the development of a worldwide navigation system. Navigation involves a knowledge of one's position and one's velocity vector. In space, these quantities are characterized by three parameters each, or six parameters in all. In purely celestial navigation the various heavenly bodies are recognized with optical instruments as indicators to determine one's position and velocity.

### Absolute Position

Observation of the stars is not sufficient to obtain positional information because the stars are in fact moving, but for all practical purposes they are stationary. The only information that can be gained from star observations in space is the direction of the stars vector from the origin of the coordinate system in the ship. To determine absolute position, the magnitude of the stars vector must be known. This can be done with a reasonable degree of accuracy by measuring the apparent diameter of the sun. If we assume an error of one second of arc in measuring the sun's apparent angular diameter, and further assume the ship is at a distance from the sun equal to the earth's distance (approximately 100,000,000 mi.) the resulting error in the magnitude of the stars vector is determined to be 100,000 mi.

Astrical method of navigation is to perform observations on the planets instead of the stars. This method will yield absolute position with great positional accuracy, but the stars are themselves observed, but it requires an accurate knowledge of the positions of the planets at any given time. This involves the use of a clock, absolute ship observations of the planets for accuracy within the orbit of Mars with instruments capable of accuracy of a second of arc and a clock accurate to a hundredth of a second would result, in most cases, positional error of the order of 1,000 mi. This indicates an error in celestial bodies can be quite accurate and would certainly enable the space traveler to find their destinations.

### Radar Techniques

Guidance can also be obtained by the use of radar techniques. Then, because similar to those used in radio guidance of aircraft could be established.

The principal difference between present-day beam and space beams is

that earth-based beams are fixed with respect to an earth frame of reference, but space beams will of necessity have a width with respect to the solar system's frame of reference. This complicates the problem because both the guiding beam and the guided ship move relative to the point of arrival. Nevertheless, such a beam system still could be conceived in several forms.

- **Beams aimed between planets**, forming polygons of variable size, the sides of which represent distinct geometric distances between planets.
- **Beams which cover periodically a large volume of space.**
- **Beams or semi-directional beams** which provide an intercept course and open the possibility of a preprogrammed navigation system.

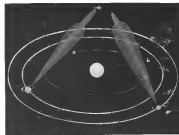
Of these possibilities, the first appears promising because it would show the spacecraft to operate in much the same fashion as a homing missile. The homing is a very convenient action for guidance, but it implies an end-to-end connection such as a laser, and therefore astronomical light paths and extremely high terminal accelerations. This line of guidance is definitely reserved for the fast spacecraft of the future.

Also, it could not guide the ship all the way to its destination planet, but would require a terminal phase of guidance because of the high terminal acceleration required.

Using the third approach, the beam would be located on the base of arrival, and the ship could navigate as much the same fashion as an intercept (pre-programmed) homing missile. This type method is considerably less the shortest straight line course to the target. In such a case the beam would need to be broader than in the previous cases, and probably it could be an omnidirectional radiating source. Navigation in this case would at least be simple in theory. The ship's pilot would select the particular frequency corresponding to the destination planet, and the automatic mechanism within the ship would do the rest, always maintaining the ship in a shortest-distance collision course.

The source of the beam could be located on the planet of arrival, or on a satellite of the planet—either artificial or natural. Because of the nature of most of all these bodies, the beam would have to be reflected or retransmitted. In the case of an astronomical radiation, several would have to be installed on the surface of a planet so as to avoid entering a shadow when the beam is transmitted from the planet.

An extension of 30 ft diameter operating at a 3 centimeter wave length (X-band) could theoretically guide a ship from an Earth base to Mars at its closest distance with an accuracy of



**RADAR BEAMS**, like those used in missile guidance, might be utilized for space navigation, providing accurate guidance of vehicles to within 10,000 mi. of destination.



**SEMI-DIRECTIONAL** beams, using low semi-directional beams directed at planets or satellites whose orbits around sun are known, is another possible approach to space navigation.

30,000 mi. This compares very favorably with accuracies quoted in celestial navigation, however, it is probably an unacceptably high accuracy.

Basic problem of beam guidance is range. Range can be increased by narrowing the beam, but the beam becomes more difficult to locate by the target. An idea of this is that the beam could be directed to 2.4 billion miles, satisfactory for traffic within the orbit of Uranus. Even as modified navigators equipped with maps and optical instruments would be able to effect a crossing

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with the beam, and then be guided by the beam.

It should be noted that use of the beam will add a direct measurement of the ship's velocity with respect to the source of the transmitted energy by exploiting the doppler effect, provided the frequency of the beam is known.

Another important system employs beams and is similar to the previously cited Loran (long-range navigation) system, except that Loran operates in two dimensions and the space range system operates in three. (See sketch, p. 57) Thus, at least four coast directional transmitting beacons could be set up in known orbits around the sea. Portion of the ship within the Solar System could be determined by measuring the difference in times of arrival of pulses issued by the four beacons, which would have to transmit pulses synchronous in time similar to the radio system.

Difficulties for space navigation is the requirement that this system be one-dimensional bearing range composed with directional beacons. An example: a directional beacon transmitting an energy power of 10 kw, used with a 10 mi. ft. receiving antenna and a good receiver displays a reliable range of about 10 miles miles. This range is far too limited for interplanetary flight.

### Better Techniques

Future advances in communications techniques should require this system. For example, it is known that the most powerful reflex is a source that is able to discriminate between signals and external noise through coding its symbols. It is also known that the system generated receiver must be refined when the transmission of components is isolated. Making use of the very low temperatures of space, it should be possible to read a message without a chain of three-fold increase in range.

An additional advantage may be gained from the fact that interplanetary distances being very large compared to the speeds there is a greater length of time available to observe the signal. This situation presents an advantage over antennas earth communications where time is limited to the high speed of signals with respect to the distance traveled. The longer the time to observe a signal, the better the signal can be distinguished over noise.

To utilize the advantage fully, certain techniques are required which have not been experimented. If a time span of one minute is allowed for observation, reference is needed, and successful use is fully exploited, range can be increased to 100 miles or more. This range would be suitable for navigation within the orbit the Air Corps.

It is difficult at the present time to

predict what advances in information theory may be expected, however, in terms of an additional factor of 10 and too far ahead. This would have that increase range to 1 billion miles which satisfaction for navigation within the orbit of Saturn. Even without such expansion, the further into space one goes the longer the time required for the trip and consequently the time for observation.

Three stage rockets approximately in the same orbit of the planet Saturn, it becomes apparent that observations of the orbit of 10 minutes could yield ranges of 1 billion miles. Besides, Solar System planet orbits all lie approximately in a plane. Thus, space navigation within the Solar System can still be confined within the plane unless other astronomical objects are discovered in other areas of space, as which possibility is the light of our present knowledge.

### Known Sector

Now the beacon's radiation pattern need be one-dimensional in azimuth but confined to a relatively narrow direction sector further increasing the beacon's range. If the virtual radiation be confined to a 10 deg sector, a range of about 4 billion mi could be achieved with 1975 technology and with a 10 minute signal observation time.

It appears reasonable to predict that such a navigational system employing beacons will be feasible in the near future and is a possibility. The known system will enable the spacecraft to determine not only its position, but also its velocity vector by making use of the doppler frequency shift. The receiver would have four different frequency spectrometers and be equipped with automatic means to measure the difference in arrival time of the four pulses and thus compute the position of the spacecraft within the Solar System. Use of the doppler technique will also enable the ship to compute automatically its velocity vector. And as the ship can be continuously located in space.

It is possible, and perhaps desirable, to locate the beacons as planets as on satellites which already possess such known orbits.

Next problem facing the space traveler in communication, time-to-ship and ship-to-ship, time-to-ship, communication, has the advantage that a very powerful station can be set up on the base for transmission to the ship.

Assuming the base to be equipped with an omnidirectional transmitter with a power output of 100 kw, and that the space ship gained a 10 mi. ft. receiving antenna, single sideband long range transmissions should have a range in the order of 4.2 million mi.

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would help since range is proportional to the square root of the power, hence, the "best bet" would be to acquire low intensities with power of all proportions. For example, if transmitter power were increased to 2 million watts, the range would only increase by a factor of 0.5, yielding a total range of 25 million miles.

When one considers that the most powerful radio transmission in the world today has been about 1 megawatt, one can visualize the tremendous powers required for such communication.

Conventional communications, however, can be improved considerably by using beacon communications in which the transmitted power is sent out as a beam and reaches only the ships that are in the beam path. What is gained is increased range, it has in increased coverage. By using a 24 deg. beam, and reaching to the 100 kw. power of the example, a range of 62 million miles could be achieved in 250 million miles with one attempt.

Beacon type communications require either a knowledge of where the ship is, or in regard to keeping the beam of power where the ship is thought to be, until the ship is found. A great deal of work in this range can be obtained as previously explained, by cooling the receiver, but the factor of extended time reception will not be as favorable. In some communications, it is desired to convey the message immediately.

### Coded Communications

Telegraphic or code communications could wait for extended periods of time to achieve increased range. In this latter case, communications would be agreed out in time, i.e., a message which would normally take one minute to deliver would require a considerably longer interval. If a continuous message is spaced out to one hour, an eight fold increase in range can be obtained. A large antenna could be erected from the ship to capture some of the transmitted energy. In this case, area of perhaps 100 square meters could be achieved with a theoretical increase in range over the previous example by a factor of 10. With the cooling technique plus the antenna increase technique, the range for communications could be increased to about 600 million miles with one megawatt transmitted power, which is a rather respectable figure.

Actually the ship is probably known to be in one area such as a consequent increase in range of a factor of 2.7, yielding a total range of about 1.6 billion miles. Again it is quite possible that future advances in technology may further extend the range. The situation for communications from the ship to the base is somewhat different. Unless future advances in

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**Gyro Stable Platform, Type DGP-9 (Polarization Platform)**  
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**Rate Measuring Gyroscopes, Type R for guided missile control and homing systems and flight systems of military aircraft.** These gyros measure angular velocity about either one or both axes. They are rugged and accurate dynamic components are available. Laser output signal proportional to input rate, within 0.25% of full scale. Withstand 1000 shock in any plane and 150 vibrations up to 2000 cps. Weight — 100 lbs. Size — 3 1/2" long x 3 1/2" diameter. Price for Model R: \$1



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with as limited power but with a high level of accuracy and direct to output. The transmitter beam would then be broadcast the message of high power.

This situation would require lead to problems of accuracy of transmitter beam, in spite of the precision. Transmitter beam would become a major obstacle, eventually only be used for (unreliable) message. Limiting accuracy in communications between the usually close spacecraft, is to see what can be achieved.

Assuming omnidirectional broadcast, a 10 sq ft antenna one kilometer of transmitted power, and very close system, a range of 0.63 radian angle can be achieved. With cooling of the receiver, the range can be increased to about 1.6 million miles. These parameters define a wave length of about 1.6 meters corresponding to a frequency of the order of 200 megacycles. An increase in range can be had by increase in the transmitted power. If future satellites will enable this ship to carry a 100 kw transmitter, a range of 1.6 million miles could be achieved.

### Beam Techniques

Another considerable improvement can be obtained by employing beam techniques. Of course the beam techniques apply a knowledge of the position of the other ship. If there is no such knowledge, systems must be had to scanning, which means searching microwave regions in space and listening for a reply.

Scanning would be complicated in space because a considerable delay between transmission and reception would be experienced. The speed of light being 186,000 mi per second, an interplanetary distance of 16 million miles is that between Earth and Mars at their closest point results in a delay of about 10 minutes. Thus a complete round trip transmission between two space ships 16 million miles apart would take close to 20 minutes.

In order to maintain contact, the transmitting ship would have to wait for detection at a complete round trip of a transmission plus a reasonable length of time to ensure that the wave has not been lost. However, this can be solved by employing a number of the accurate antenna positions. In this manner the antenna could transmit its message on, for one to two minutes in each direction, then wait all to a new direction. If, so or eight minutes later, a received message appeared, it would only be necessary to have a reasonable antenna position with respect to the direction of the transmitting ship to allow the antenna beam.

It should be noted that scanning will involve higher frequencies than the HF and must likely it will require power into the VHF or microwave region to

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Alt. 400 CPS	Alt. 400 CPS	Alt. 400 CPS	Alt. 400 CPS	Alt. 400 CPS	Alt. 400 CPS	Alt. 400 CPS	
70001-1	10	4	31	6.25	100-1000	40	400-1000-400 A
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70001-3	10	4	14	6.25	100-1000	40	400-1000-400 A
70001-4	10	4	14	6.25	100-1000	40	400-1000-400 A
70001-5	10	4	14	6.25	100-1000	40	400-1000-400 A
70001-6	10	4	14	6.25	100-1000	40	400-1000-400 A
70001-7	10	4	14	6.25	100-1000	40	400-1000-400 A
70001-8	10	4	14	6.25	100-1000	40	400-1000-400 A
70001-9	10	4	14	6.25	100-1000	40	400-1000-400 A
70001-10	10	4	14	6.25	100-1000	40	400-1000-400 A
70001-11	10	4	14	6.25	100-1000	40	400-1000-400 A
70001-12	10	4	14	6.25	100-1000	40	400-1000-400 A
70001-13	10	4	14	6.25	100-1000	40	400-1000-400 A
70001-14	10	4	14	6.25	100-1000	40	400-1000-400 A
70001-15	10	4	14	6.25	100-1000	40	400-1000-400 A
70001-16	10	4	14	6.25	100-1000	40	400-1000-400 A
70001-17	10	4	14	6.25	100-1000	40	400-1000-400 A
70001-18	10	4	14	6.25	100-1000	40	400-1000-400 A
70001-19	10	4	14	6.25	100-1000	40	400-1000-400 A
70001-20	10	4	14	6.25	100-1000	40	400-1000-400 A

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low a figure of 5 db (meaning a loss of 41). It should be pointed out that this figure is considerable today at these frequencies, but may be worked within the next 10 to 20 years. Today the way lost that could probably be achieved at this frequency is 15 db, which means a loss of 30 to 1.

In order to be reliable, the signal must exceed the receiver noise by several times. This is particularly true in automatic radar, in the sense that, as an automatic radar must have a very high probability of not losing the target with random noise. Again, we will assume that in order to be reliable, the signal must exceed the noise by at least 6 db (three of four). This again is not achievable today, but can be worked in the probable future.

We now have all the elements to compute the range by applying the radar range equation. We find that the range of this radar is only 25 mi. Some things can be done now to improve this range. First we can use special circuits to increase at least partially the loss caused by the noise of the receiver. Assuming that this loss can be removed completely, we find that the range can be improved to approximately 60 mi.

An additional improvement can be derived by making the receiver, by the use of side, the improvement is not proportional to the square root of the cooling factor as in the case of two receivers, but only to the 4th root because of the two sets of receivers. Assuming that the receiver can be cooled to 70 deg. absolute, an improvement of about 2.5 results, yielding a range of approximately 150 miles.

## **Time Is Short**

Now a noticeable tracking head-on towards the ship at relative speeds of 40 mi. per second, this range gives in about 5 seconds to detect the target, plot its course, determine that it is a hostile force, compute the ship's position to guide the battery, and so on. The measurements traveling head-on to the ship at speeds of 20 mi. per second, the time to do all this is increased to about 5 seconds, a very short time to perform all these functions.

The situation is a better case than the one assuming that automatic pass the ship. If the ship should travel at a speed equal to an average automatic speed, then it will have the time once behind and will only be concerned in the latter case. These data indicate that probably will not exceed the speed of the average automatic is a few miles per second. That gives a possible time up to 1 min. to perform automatic search.

Thus in regions of space where the automatic search and speed are known, it is possible to protect against them by tracking in their direction and, of course, at different speeds. Where the

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APL is responsible for technical direction of the Navy's Langley guided missile program. Development at APL includes the first supersonic cruise, and the jet engines TERRIER, TALON and TARTAN.

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direction and speed of the velocities is unknown, it appears that the only self-producible instability in jet engines is some major adverse error in the air.

We have assumed small targets of the order of a tenth of a square foot. Against larger targets the situation is only slightly better. Since range increases as the frontal area of the target increases so that even if we increase the target size to a factor of 100 giving us a 10 sq. ft. target, which corresponds to a 1000 ft. distance, our range would only go up a little over 3 times. This increased range would be on the order of 300 mi. For head-on encounters, a 100 sq. ft. target would give us a warning time of about 5 seconds which still appears too short for all the things that must be done.

In view of the complexity of space collisions between ships appear at first sight impossible unless we assume a very high traffic density confined to relatively narrow traffic lanes. Between ships, a collision warning system could be used which would have much greater range than the radar. Such a warning system could have the approximate range calculated previously for the case of ship-to-ship encounters and then even a relatively poor ship-to-ship warning system could yield ranges of tens of thousands of miles.

### Endeavor Mapping

Plasma mapping is of course best done by optical means. Only in cases where plasma are coated by clouds other phenomena or topography, such as Venus, would radar mapping be desirable and worthwhile. In addition to mapping, radar technology can be used to some extent to detect the existence of a planet's atmosphere. Different gases or combinations of gases pose different absorption characteristics at different frequencies. One could then gain an insight into the composition of a planet's atmosphere by measuring the weakening of the radar echo from a fixed point on the planet's surface as the radar frequency is varied.

An even better method would be the measuring of a radar return transponder which could be parachuted to the planet's surface. This device would be equipped with a constant power beacon transmitter which would retransmit the frequency with which it is being interrogated by the ship.

A weakening of the returned signal at a wavelength of 1.35 centimeters (9.6 cm), for instance, would indicate the presence of water vapor. A strong absorption at 5 millimeter wavelength would detect the presence of oxygen. The amount of weakening would reveal the content of oxygen in the planet's atmosphere. This beacon technique should be well developed by 1970.



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FULL-SCALE MOCKUP of Firebird M-1500 jet transport is available for customer inspection at Hagglund, MI. Long nose is designed to house cabin. Forward door is on left ahead of the wing. Main loading area extends into the side of the cabin.

## Fairchild Sees Market for 600 M-185Fs

By Kevin J. Hoffman

New York—A market of approximately 600 units exists for the new Fordchord M-1500 four- or six- passenger bus and van, transport, according to Richard S. Boudelle, president of Fordchord Engage & Automotive Corp., Gaithersburg, Md.

Bastelli said that the Egon was developed during a market survey by the company aimed at finding out the sales potential for its new model. He is

ports that this potential will be reached within six years after delivery of the East M-15SE.

Number one production M-35IF will be delivered to Continental Can Co. sometime in 1968. The corporation has ordered three M-35IFs (AW Aug. 27, p. 35), in addition to a turboprop-powered Fairchild F-27 Friendship executive plane which it expects to receive in July 1968.

Continental Can Co., which bought

The first corporations to sign a letter of intent to purchase jet transports, will get production airplanes within one,

these involve. Fairchild will reuse the experimental prototype for development and demonstration purposes. Continued work will keep the M159 at its Muskegon, N.J., operations and maintenance base. The company now has a fleet of five prototype engine testbeds; phones there: Lockheed Lodestars - Martin R 35 Muskegon and a Corvus LB 10 Lubecite. It sets a target of 18 hr/month utilization per engine and, generally, goes above that.

### Second and Low Cost

Contractual space for guests tends to take advantage more effectively of parents' productive capacity, estimated at north thousands of dollars per hour, was indicated by company chairman Gen. Leroy D. Cline (ret.), who noted that the M-105F jets will be twice as fast as its current parent fleet. "Another important consideration is the cost factor—we estimate that the M-105F's operating cost will be about the same as part of our business-line fleet," he said.

Address of jet equipment will make the following schedules possible for complete evacuations, according to first class New York to Chicago, 1 hr. 10 min.; New York to Miami, 2 hr. 30 min.; Chicago to Dallas, 1 hr. 40 min.; Dallas to Los Angeles, 2 hr. 40 min.; Los Angeles to Kansas City, 2 hr. 30 min.

Sales reflect the long-term bet on the premium in the drinking of Stone Bros. Continental chief pilot who told Aviation Week, the new jet will have a three-year climb of 1,700

ft./min to 3,800 ft./min depending on loading. Cutting of approximately 15,000 ft can be maintained in cut at one of the turbojets flames out, he said. Operations operating altitude for the M-155T will be 45,000 ft., maintaining a cruise speed of about 470 ft. The M-255T is designed for a range of approximately 2,500 nautical miles at cruise setting taking into account a 75 mph headwind and losing 10 min. fuel reserve. Latter figure are based at sea level, at optimum altitude, the fuel reserve would extend to about 45 min.

### Performance Stretch

Continental's airplanes will be fitted with external auxiliary fuel tanks outboard of the engines giving a total of 1,800-1,900 gal.

Although a firm price has not been disclosed for the executive M-201F, it is understood to be less than \$1 million. Some certain equipment such as pilot in radio. A high Eurochild owner estimated the company will break even on sales of less than 100 airplanes. It has spent approximately \$2 million on development to date. An official estimates the program will cost about \$8 million up to the first flight of the prototype. Eurochild is financing the program in

The company will borrow four F4U-1B J85 engines with 50 hp, on three low USAF doc prototype thing. These 2,300-hr thrust class engines are expected to be standard powerplant, although the General Electric J85, an approximately the same thrust class, may be substituted. Final powerplant decision will not upon availability of engines in sufficient quantity and USAF's willingness

to release them for cockpit use. A Lockheed official noted that should JSTs be available in time for flight of the prototype, two Weibull-type J36s may be used to pass aerodynamic and flight data pending receipt of the specified engines. The prototype M-200 is expected to make its first flight late next

**Performance Stretch**  
Countdown's performance stretch is being built into the new Fairchild, an official estimate that within five years after the powerplants are certified their thrust ratings will be 25% higher than current power. Fuel economy also makes it possible to consider engines of larger dimensions than current power plants. Previous M155 configurations which had the engines mounted in the fuselage would have made this more

its causes, the same, which operates in  
the conditions and production.

Contractual Chief Pilot Bruce noted that his company has not yet made up a detailed specification of its equipment requirements in the airplane. Most of this equipment will be new to us," he noted, "maybe some of it isn't even made yet." He pointed out that "you don't put piston-engine equipment in a jet airplane." For example, the company will obtain for its M-119Fs either the Sperry or Bendix autopilots similar to those found on the Boeing 767 and Douglas DC-8. It also plans to install Bendix C-100 head warning radar.

Interior design and furnishings are other matters the corporation has not yet decided upon. Brown said that he expects to discuss these with Firebird people soon.

### All-Metal Polish Trainer Details Are Revealed

Vienna, Austria—First details to be disclosed at the next two-phase Euro air meet, shown for the first time at the Paris Air, reveal that it has a 530-hp. engine and a top speed of 390 mph.

The East Indian closely resembles the North American T-28 in general configuration (AW July 9, p. 37). Gross weight is 1,403 lb. and range is about 435 mi. Rate of climb is reported to be 21.5 ft./sec. and ceiling is 19,685 ft.

The engine is a quantity producer. Equipment and other components are installed in sub-assemblies, which are later joined on a final assembly line.



DISPLAY MODEL shows how wing leading edge is swept back about 15 deg., trailing edge about eight degrees. Extra fuel tanks will be hung between the engines and the reaction



### Aeroflot Yak-12 Handles Light Utility Roles

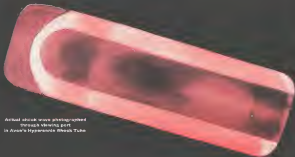
Yak-12F, nicknamed Galeb by NATO, is used by the Russians for a wide variety of utility duties including crop dusting, soil leveling and seedbed systems. This airplane has the unique feature of Autolift, Russian automatic landing, replacing the crew's device operation. Among the plane's features are climb speed over the cluster reaching the limit of the 140 kg, 1000, probably to maintain proper temperatures in duct operations, full-time loading along wing slots and long-term trailing edge flap in power flying in and out of small, unimproved fields.







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